

Experimental Identification of a Damage Model for Composites using the Grid Technique Coupled to the Virtual Fields Method

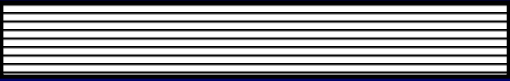
**Mr Hocine CHALAL, Dr Fodil MERAGHNI,
Dr Stéphane AVRIL, Professor Fabrice PIERRON**



LMPF Research Group
ENSAM
Châlons en Champagne
France



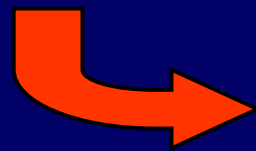
● Usual strategy for in-plane orthotropic elastic moduli measurements

←  → E_{xx}, ν_{xy}
0° tensile test

←  → E_{yy}
90° tensile test

←  → G_{xy}
shear test (off-axis)

3 tests
4 parameters

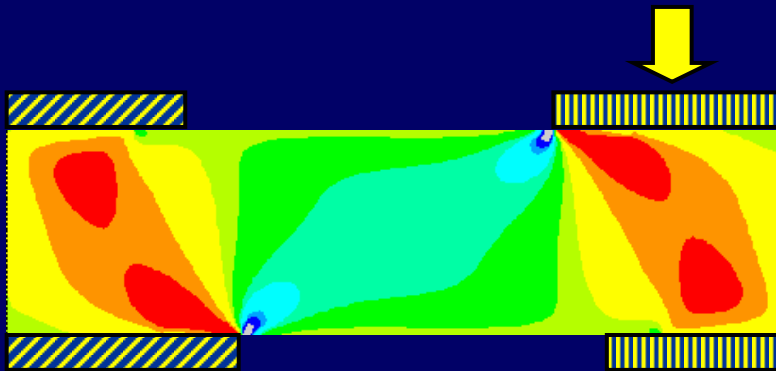


Local strain measurements

Uniform stress fields (closed-form solution)

Spatial stress distribution

● **Novel** strategy for material parameter identification

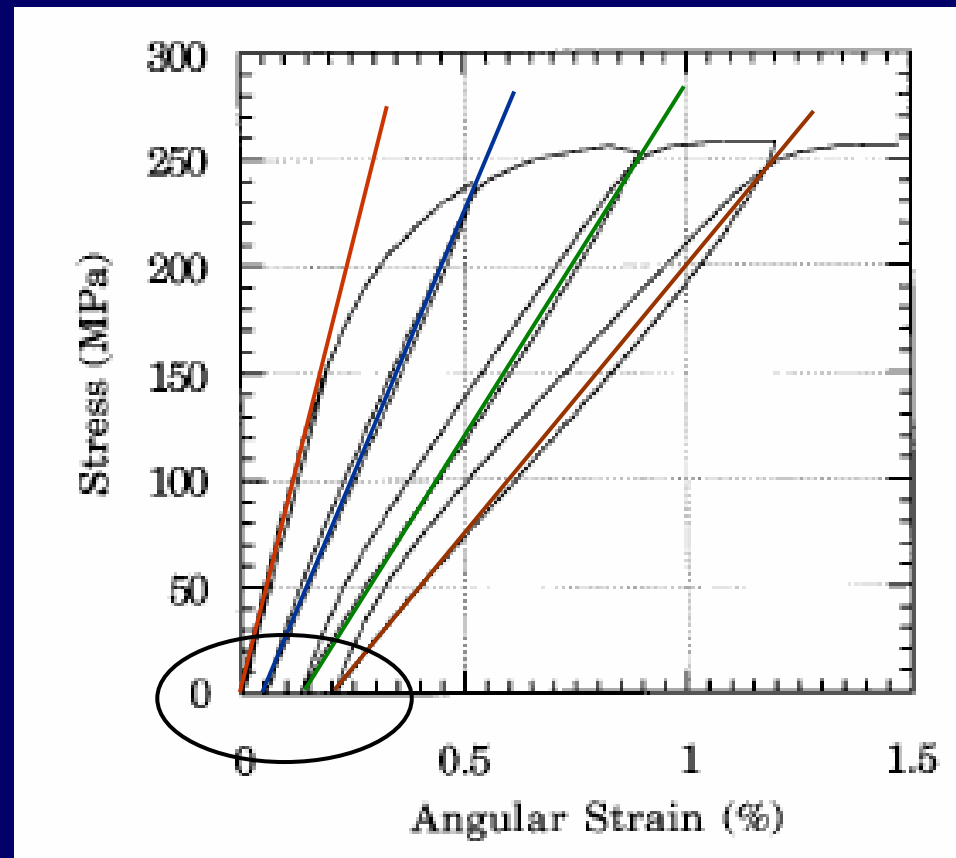
 E_{xx} E_{yy} ν_{xy} G_{xy} **1 test**

Heterogeneous stress fields
(no closed-form solution)
Full-field strain measurements

Inverse problem

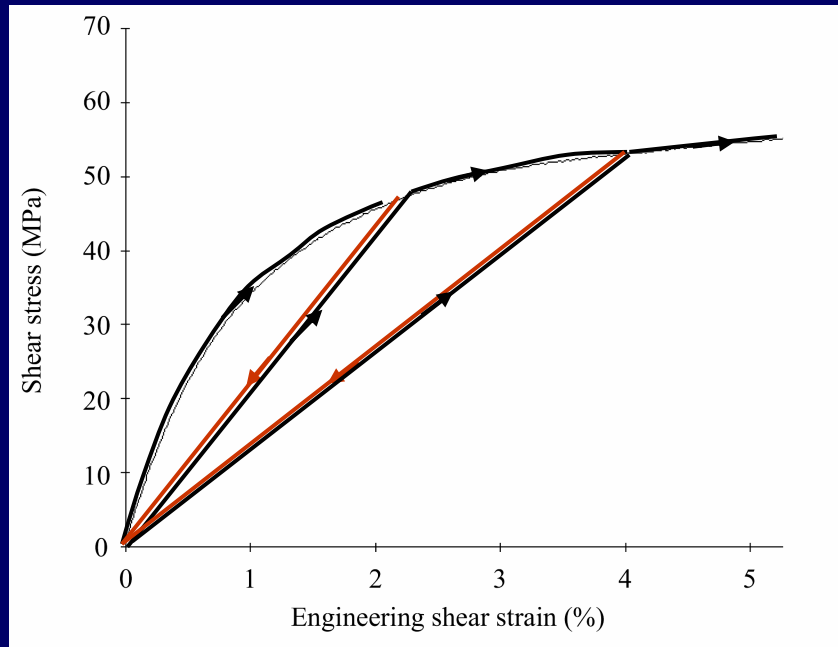
**FE model updating****Virtual Fields Method**

● In-plane shear non linearity



2D Sic/Sic composite (Camus, IJSS, 2000)

● In-plane shear non linearity



$$\sigma_s = G_{xy} \varepsilon_s - K \varepsilon_s^3$$

Damage law

$$\sigma_s = G_{xy}^0 (1 - d) \varepsilon_s$$

$$d = \frac{K}{G_{xy}^0} \varepsilon_s^2$$

Retrieve $E_{xx}, E_{yy}, \nu_{xy}, G_{xy}, K$

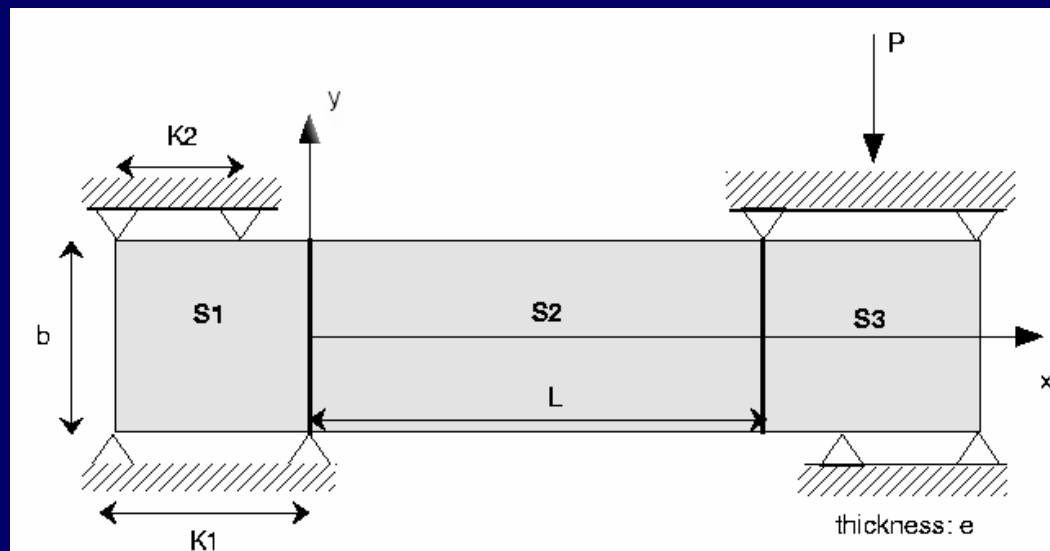
from a heterogeneous strain map

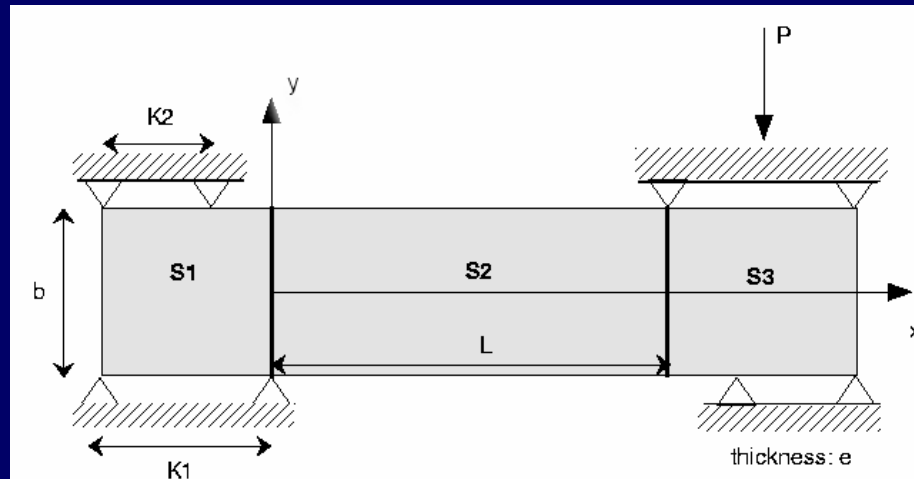
Principle

Global equilibrium of structure

➤ Principle of virtual work $-\int_V \sigma_{ij} \varepsilon_{ij}^* dV + \int_{\partial V} T u^* dS = 0$

Choice of virtual fields

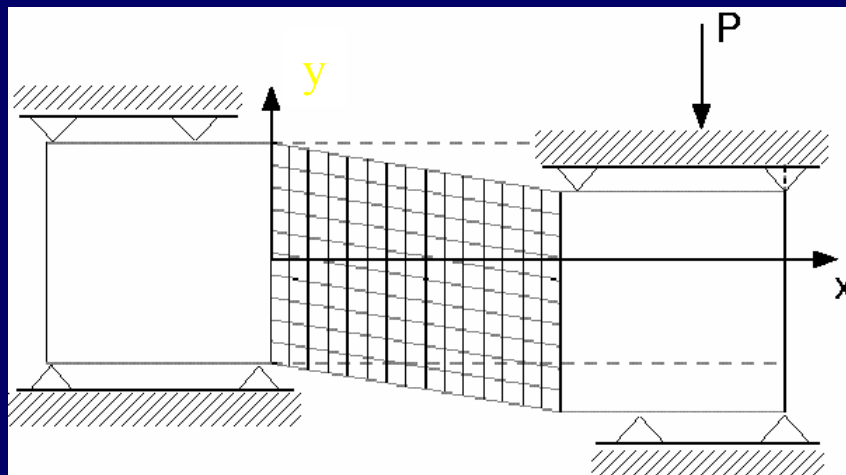




Over S2

$$u_x^* = 0; u_y^* = -x$$

$$\epsilon_x^* = 0; \epsilon_y^* = 0; \epsilon_s^* = -1$$



Uniform virtual shear

$$-\int_V \sigma_{ij} \varepsilon_{ij}^* dV + \int_{\partial V} T u^* dS = 0$$

↓ Plane stress

$$\int_V \sigma_s dV \rightarrow t \int_S \sigma_s dS$$

Material

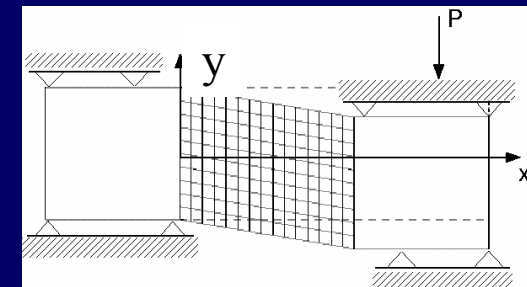
homogeneous

$$t \left(\int_S Q_{ss} \varepsilon_s dS - \int_S K \varepsilon_s^3 dS \right) \rightarrow t \left(Q_{ss} \int_S \varepsilon_s dS - K \int_S \varepsilon_s^3 dS \right)$$

In-plane elastic orthotropy
with shear damage

$$\begin{pmatrix} \sigma_x \\ \sigma_y \\ \sigma_s \end{pmatrix} = \begin{bmatrix} Q_{xx} & Q_{xy} & 0 \\ Q_{xy} & Q_{yy} & 0 \\ 0 & 0 & Q_{ss} - K \varepsilon_s^2 \end{bmatrix} \begin{pmatrix} \varepsilon_x \\ \varepsilon_y \\ \varepsilon_s \end{pmatrix}$$

$$Q_{ss} \int_{S_2} \varepsilon_s dx dy - K \int_{S_2} \varepsilon_s^3 dx dy = \frac{PL}{t}$$



-PL

Need to choose other virtual fields

$$Q_{xx} \int_{S_2} \varepsilon_x \varepsilon_x^* dx dy + Q_{yy} \int_{S_2} \varepsilon_y \varepsilon_y^* dx dy + Q_{xy} \int_{S_2} (\varepsilon_x \varepsilon_y^* + \varepsilon_y \varepsilon_x^*) dx dy \\ + Q_{ss} \int_{S_2} \varepsilon_s dx dy - K \int_{S_2} \varepsilon_s^3 dx dy = \frac{u_y^*(L)}{t}$$

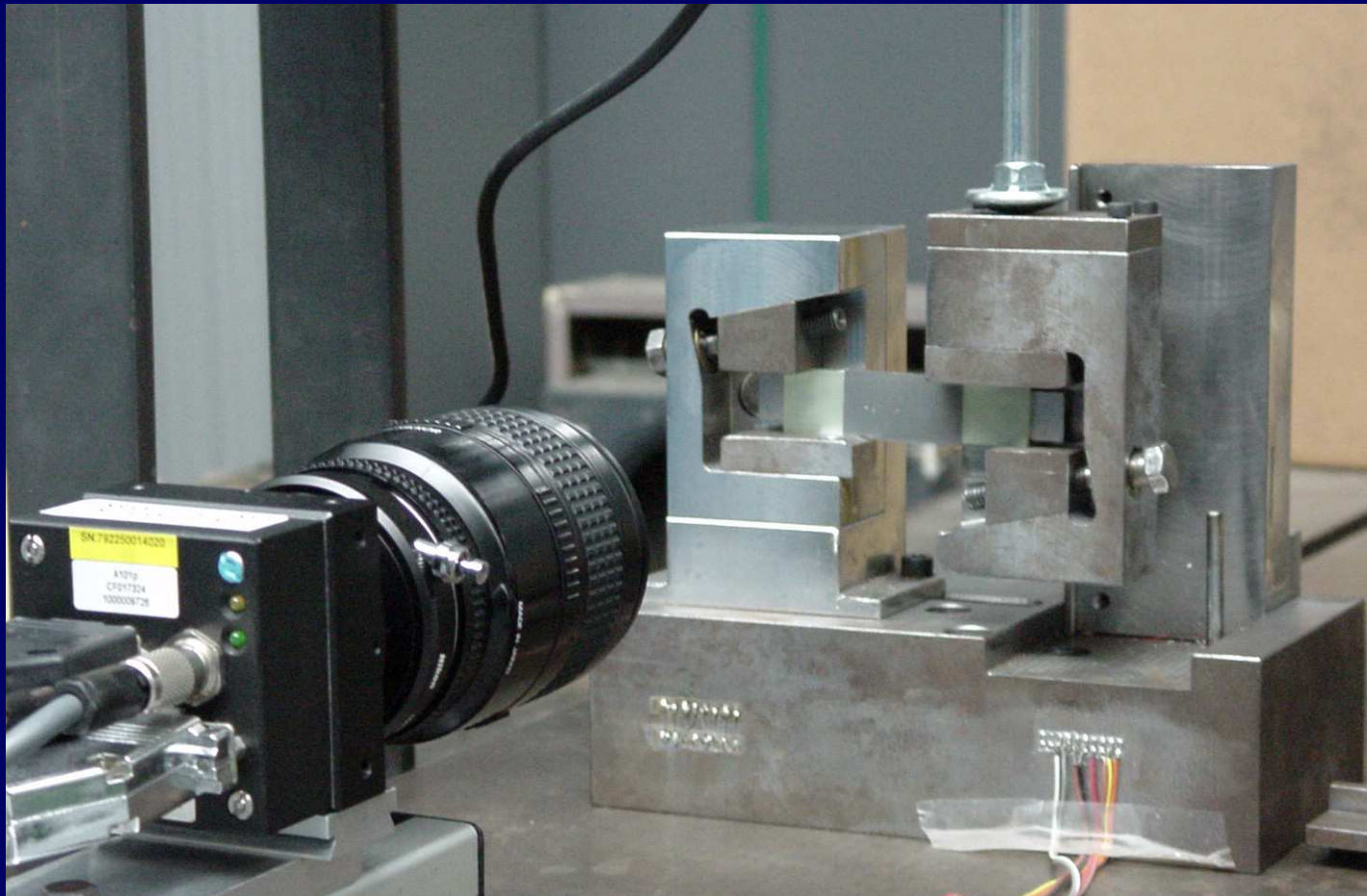
Finally

$$AQ = B \quad \Rightarrow \quad Q = A^{-1}B \quad \text{if the fields are « well chosen »}$$

Special virtual fields

$$A = I \quad \Rightarrow \quad Q = B \quad \text{optimized to resist to noise in data}$$

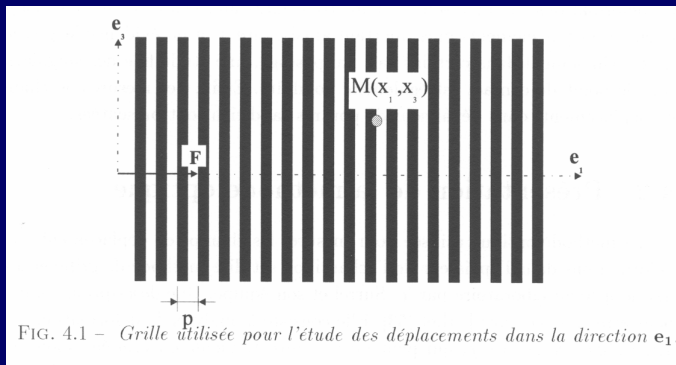
● Experimental set-up



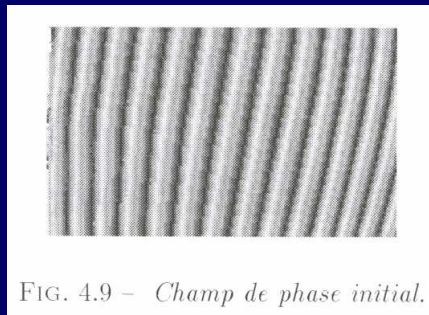
Material: 0° glass/epoxy composite (2.1 mm thick)

The grid method

Grid glued on the specimen

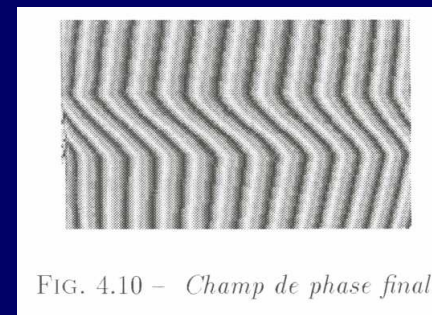
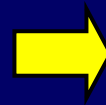


Spatial phase shifting

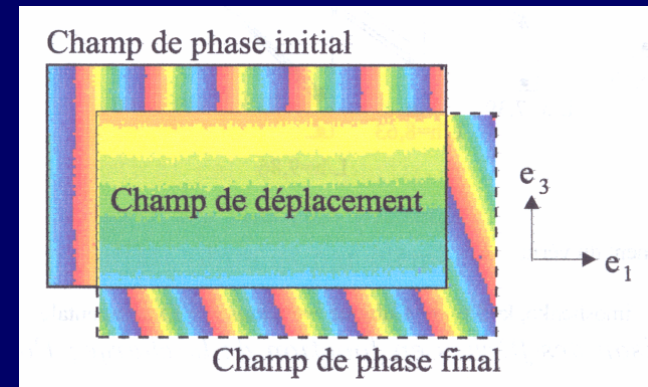


Phase map (undeformed state)

Loading



Phase map (deformed state)

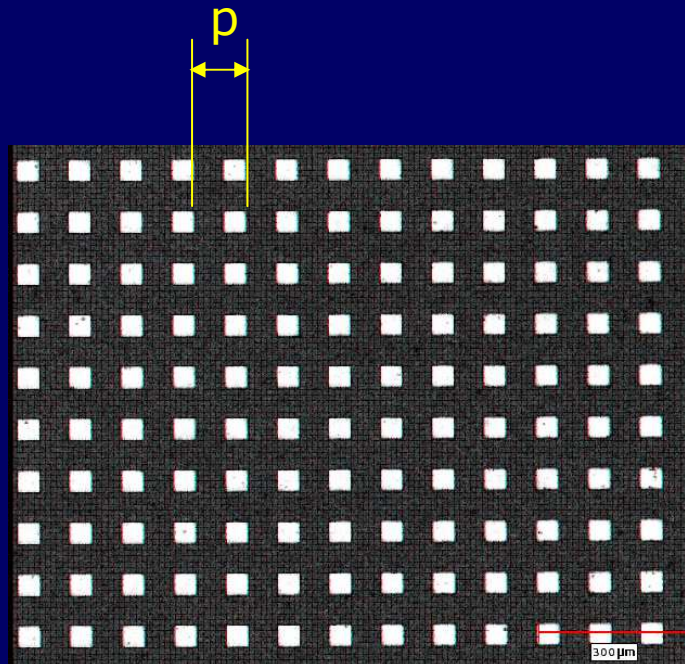


Difference

● The grid method

- ↳ Printed on a photosensitive film (postscript file)
- ↳ Photosensitive layer is transferred onto specimen (white glue), backing film removed

Cross grid of pitch $p=100\text{ }\mu\text{m}$

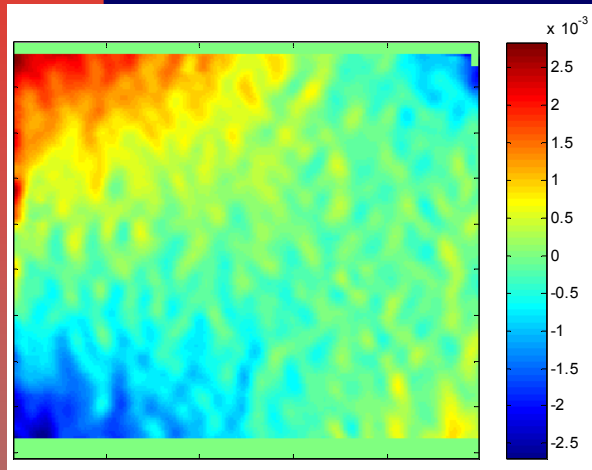
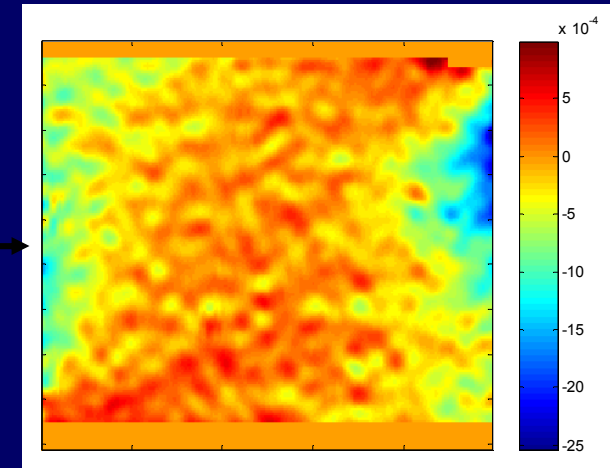
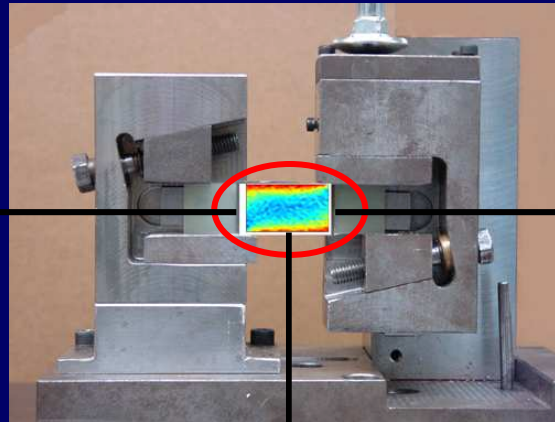
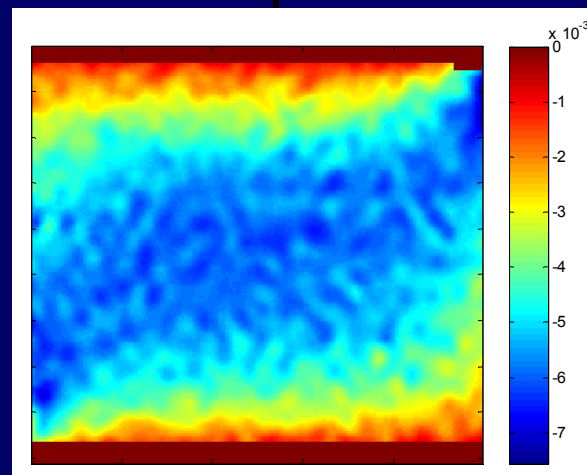


● Experimental details

- ↳ Digital CCD Camera (8bits, 1280×1024)
- ↳ 60 mm focal lens
- ↳ Frangyne software (Prof. Y. Surrel)
- ↳ Field of view: 30 mm by 20 mm
- ↳ 4 pixels /period
- ↳ Smoothing phase maps: 33 x 33, twice
- ↳ Differentiation: 33 x 33 pixels

Spatial resol. in displacement Δx	Resol. in displacement σ_u	Spatial resol. in strain $\Delta \varepsilon$	Resol. in strain σ_ε
200 μm	0.60 μm (p/170)	1.4 mm	120 microdef.

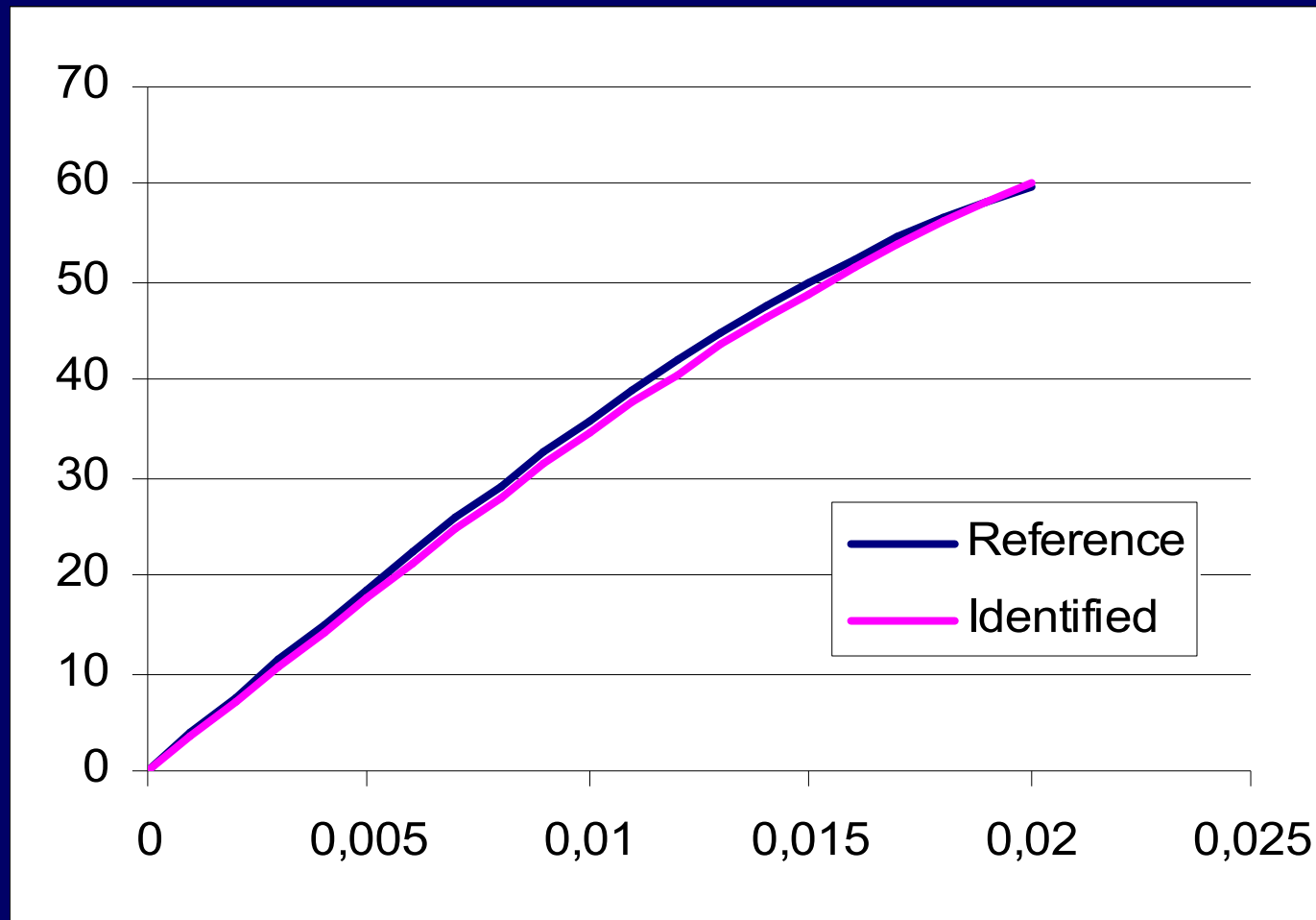
Strain maps

 ϵ_x  ϵ_y  ϵ_s

 Identification results Special virtual fields

	Q_{xx}	Q_{yy}	Q_{xy}	Q_{ss}	K
Reference (GPa)	44.9	12.2	3.86	3.78	1990
Coeff. var (%)	0.7	2.8	2.4	7.3	31
Identified (GPa)	56	27	-	3.6	1500
Rel. diff. (%)	25	130	*	-5	-25

In-plane strain response



● Conclusions

- ↳ VFM: process strain maps
- ↳ VFM: Material non-linearity
- ↳ Good results if information there
- ↳ Problem: what test?

● Perspectives

- ↳ Fibres at an angle (45° ?)
- ↳ Design of new tests adapted to VFM
- ↳ Implement more sophisticated constitutive models