



Modeling Off-Axis Notch Sensitivity of Fiber Metal Laminates

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Outline

1. Background

- Notched strengths of composites
- Objectives

2. Experiments

- Glare-3
- Off-axis tension tests on notched specimens

3. Analytical Modeling & Verification

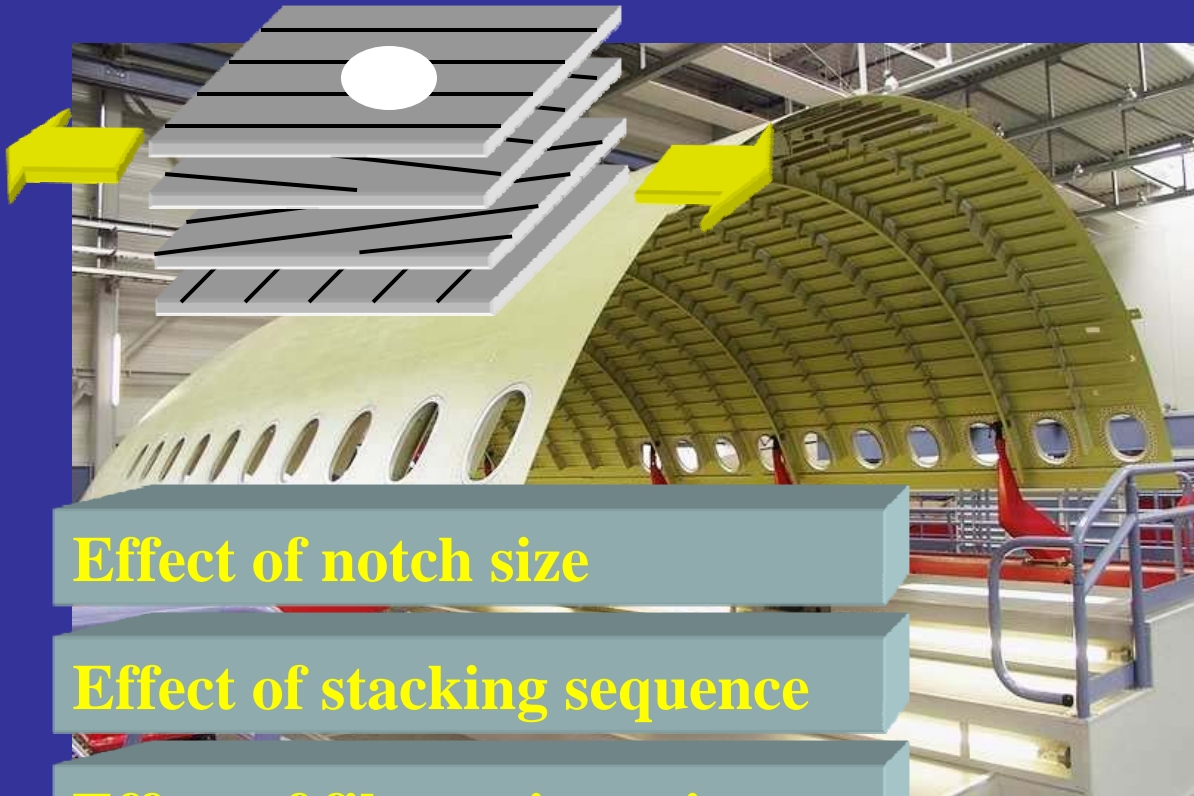
- Multiaxial notch sensitive insensitive failure criteria
- A formula for off-axis notched strength
- Comparison with experimental results

4. Conclusions



Notch Sensitivity of Composites

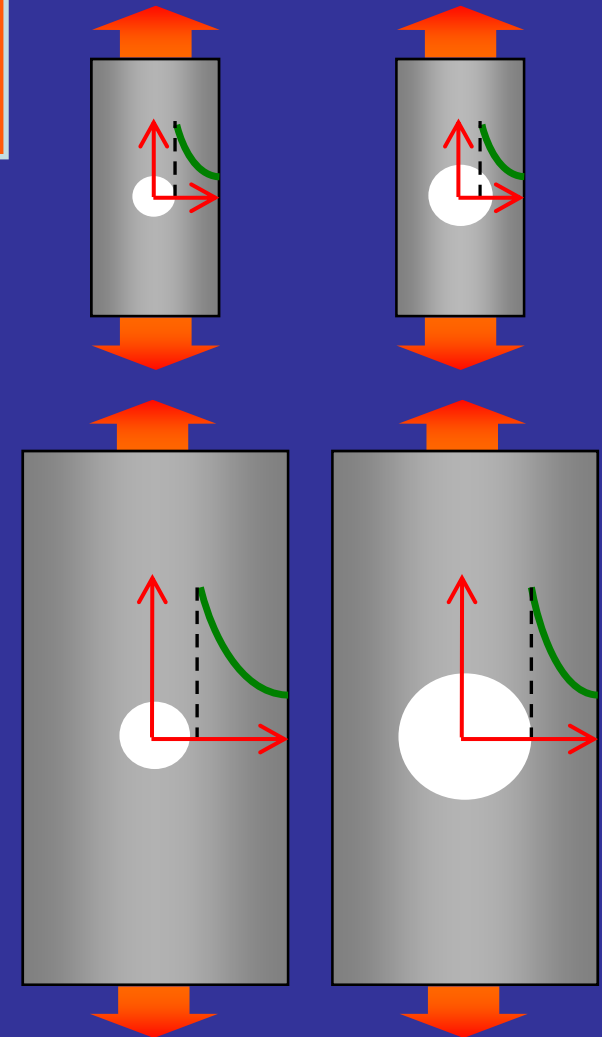
It is one of the most important engineering issues to establish a reliable method for predicting the notched strengths of composites.



Effect of notch size

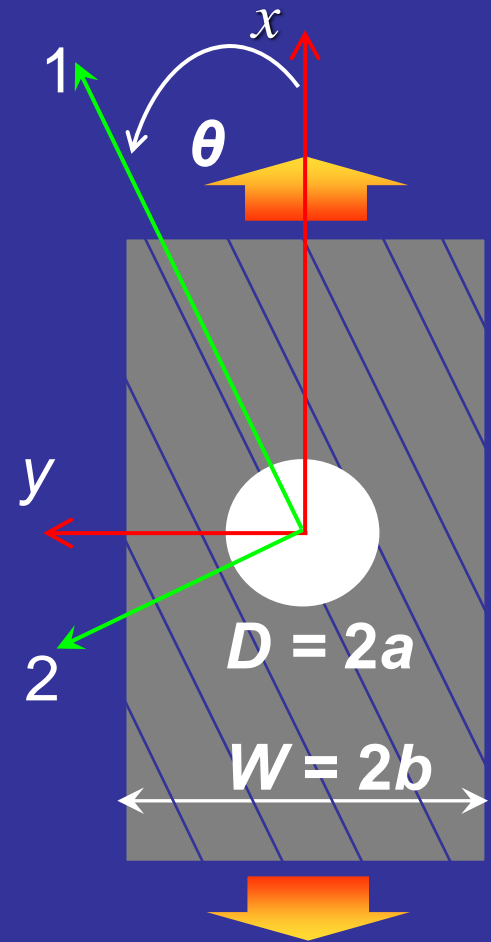
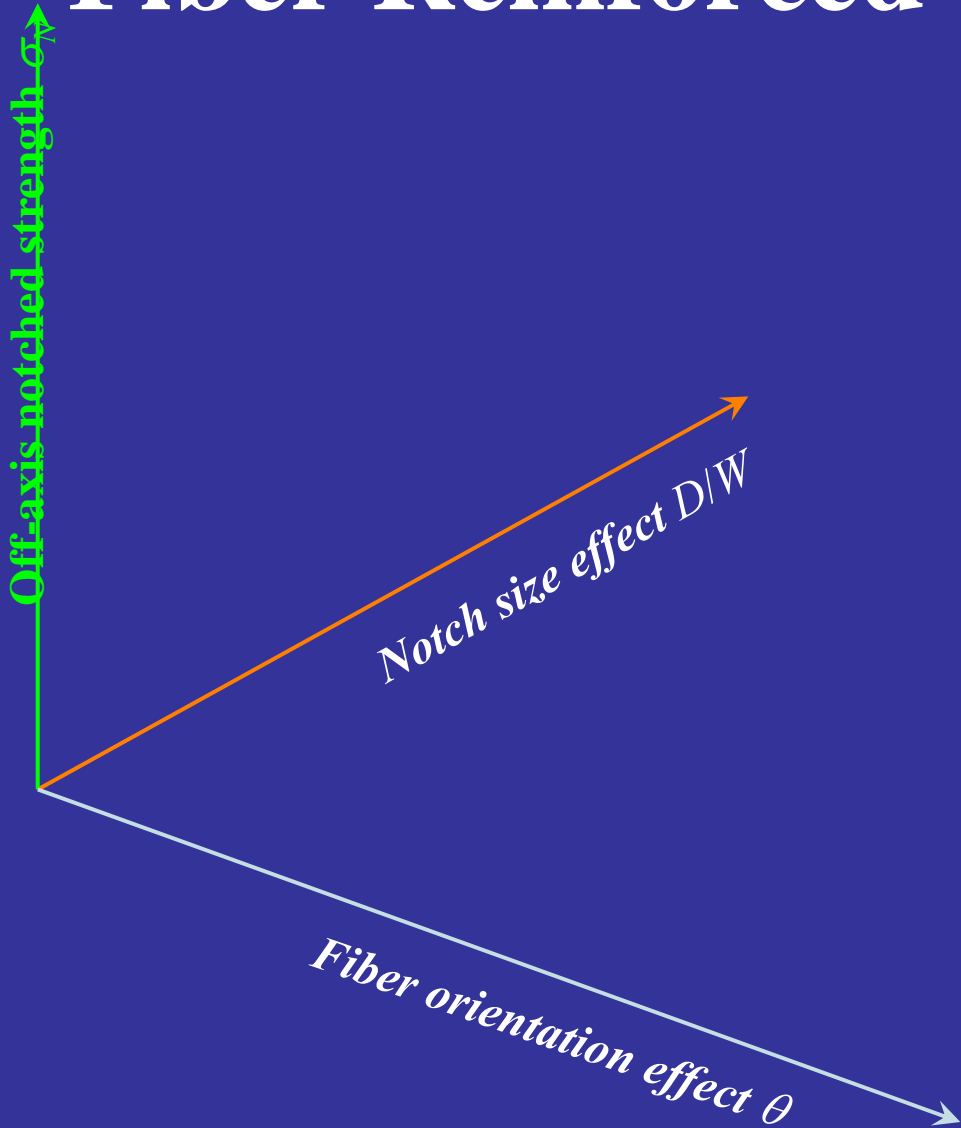
Effect of stacking sequence

Effect of fiber orientation





Off-Axis Notched Strengths of Fiber Reinforced Composites





Objectives

Modeling of the off-axis notched strengths of fiber metal laminate GLARE-3

EXPERIMENTAL:

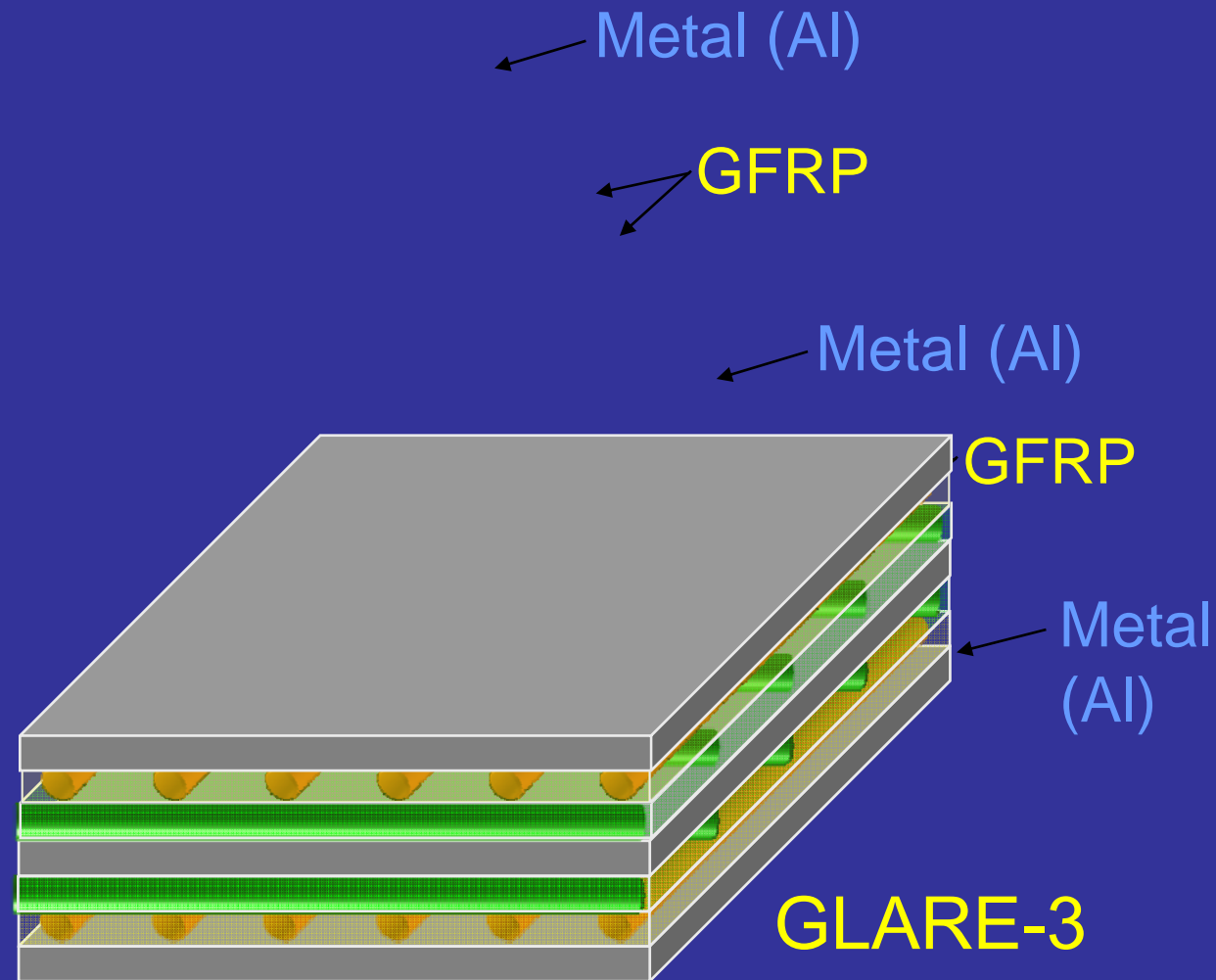
- 1 Notch size effect
- 2 Fiber orientation dependence
- 3 Off-axis notch sensitivity

THEORETICAL:

- 1 Formulation of a multiaxial failure criterion
 - A criterion for ductile failure
 - A criterion for brittle failure
 - A criterion for transitional ductile-brittle failure
- 2 A formula for off-axis notched strength prediction



FML (Fiber Metal Laminates)



► GLARE

Al alloy + GFRP (Glass Fiber Reinforced Plastic)



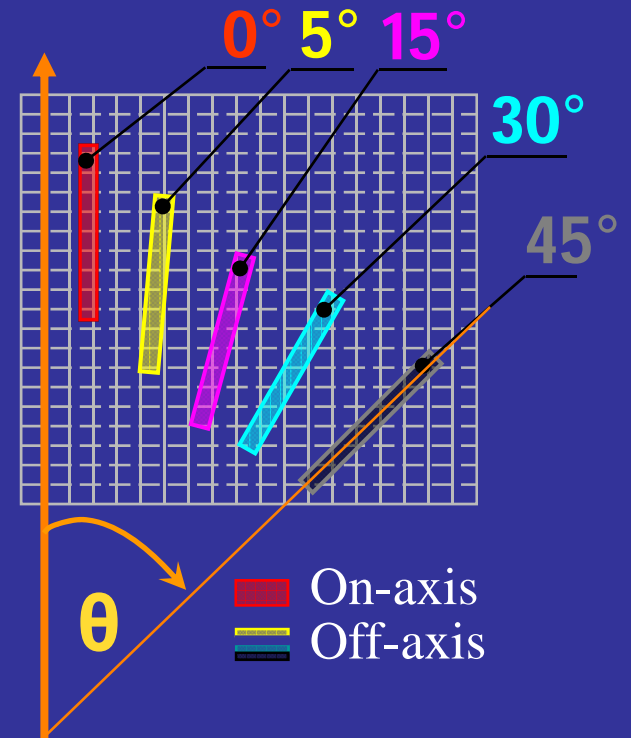
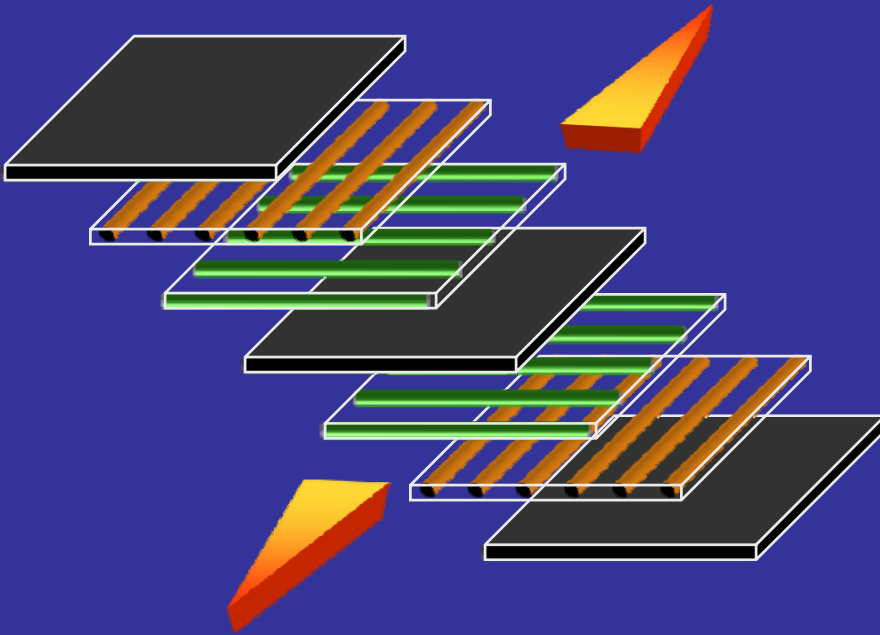
Material

▶ GLARE-3

Al alloy sheet

0°GFRP

90°GFRP

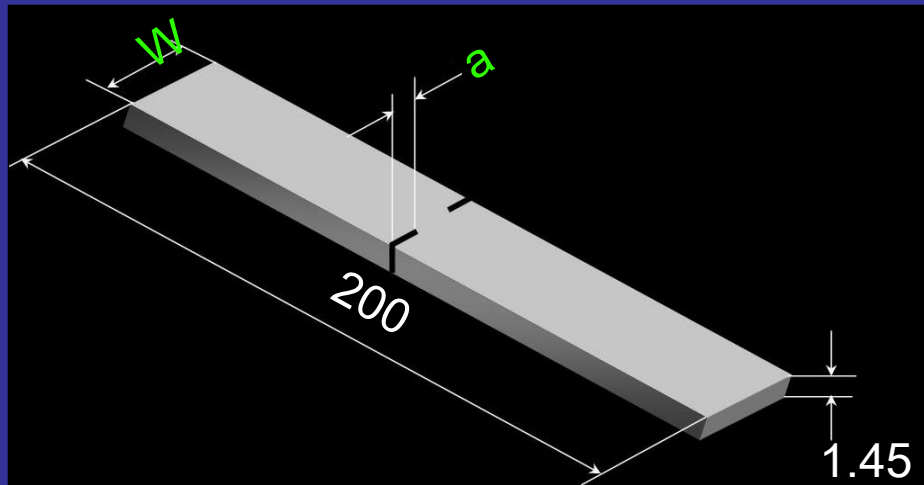




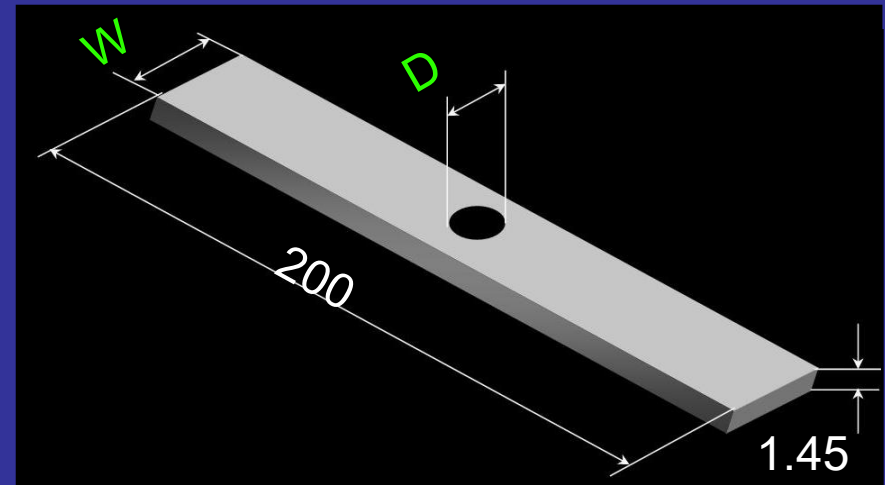
Notched Specimens

Off-axis angle	θ	: 0, 5, 15, 30, 45, 90°
Width	W	: 10, 20, 30 mm
Normalized width	D/W	: 0.0, 0.1, 0.2, 0.4
	$2a/W$:

Double edge notches



Center circular hole





Test Procedure

JIS K7073

- » Temperature
RT
- » Rate
1.0 mm/min
- » Strain Measurement
Strain Gauge
DVE
Disp
Extensometer



MTS Test Star 810



Extensometer

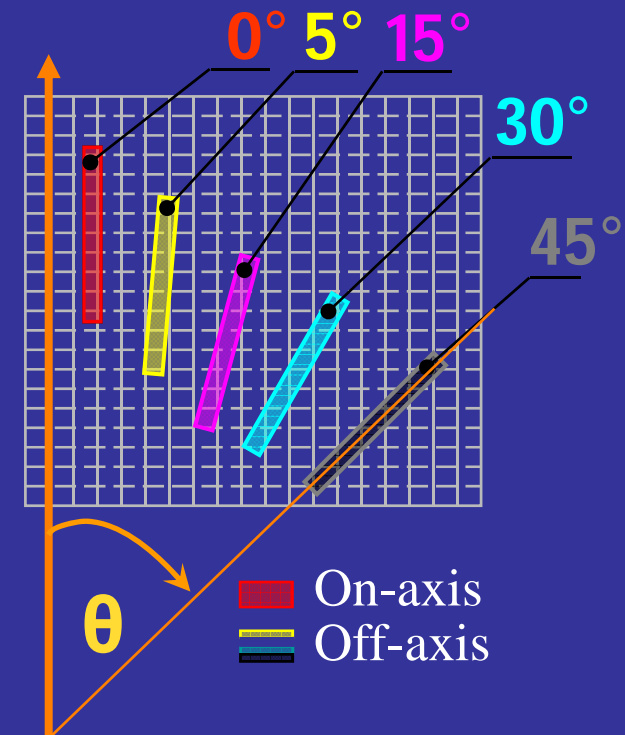
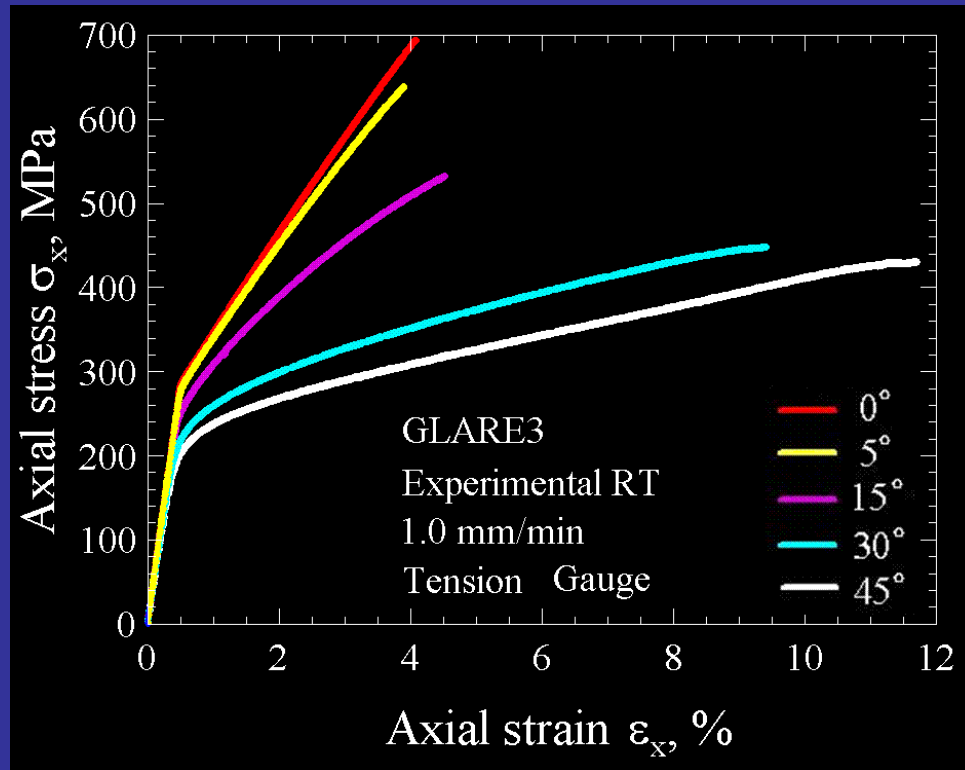


Experimental Results



Off-axis Stress-Strain Curves

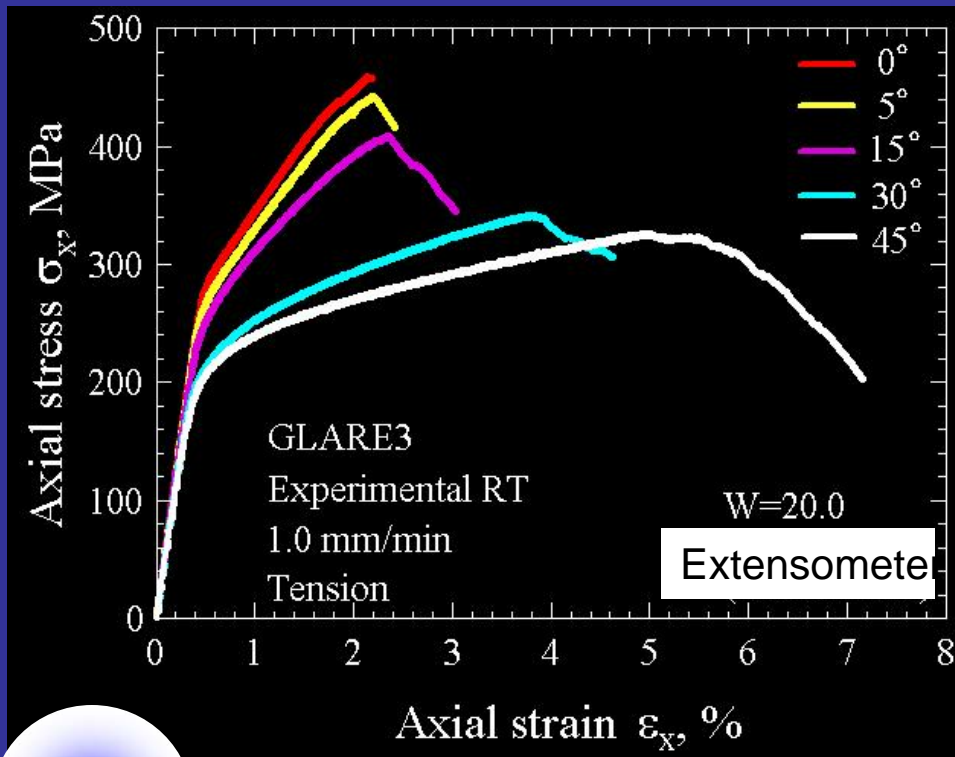
Unnotched behavior



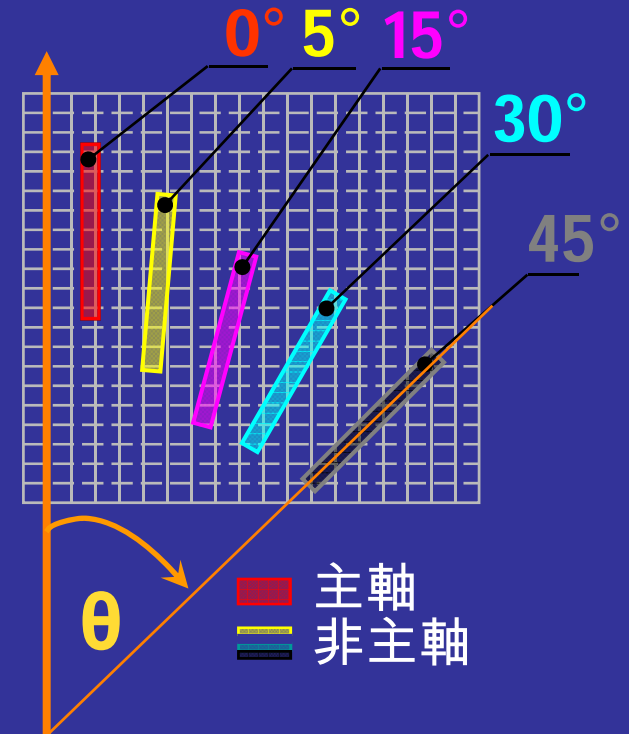
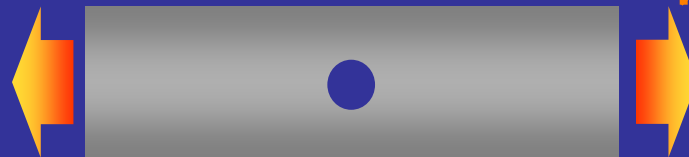


Off-Axis Stress-Strain Curves

Notched behavior



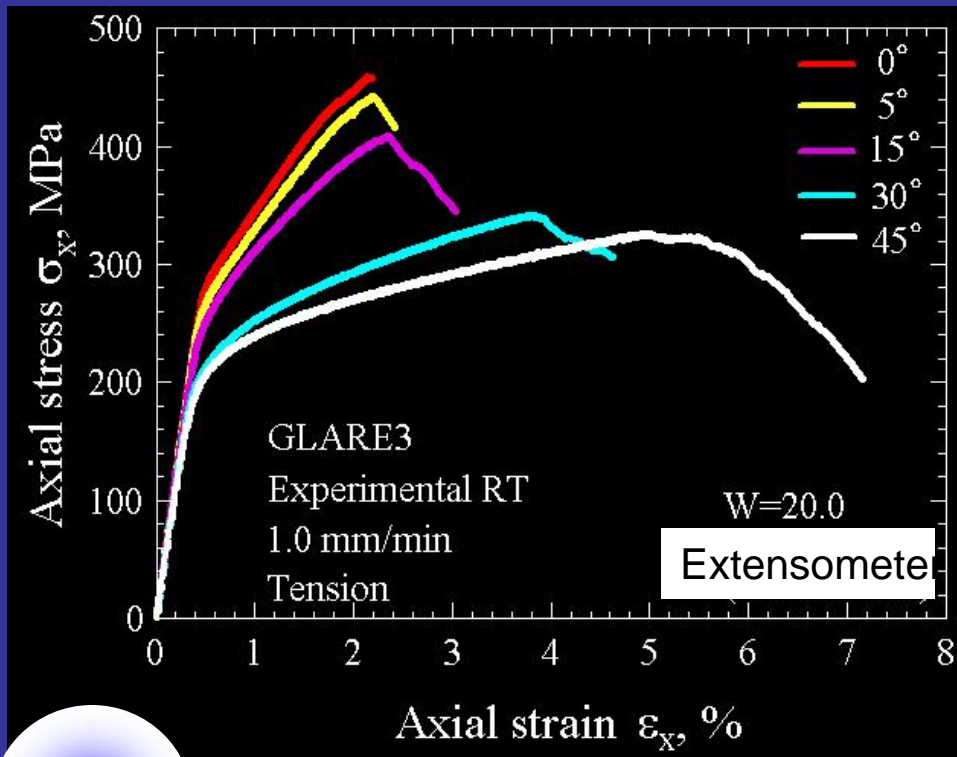
$D = 2.0$



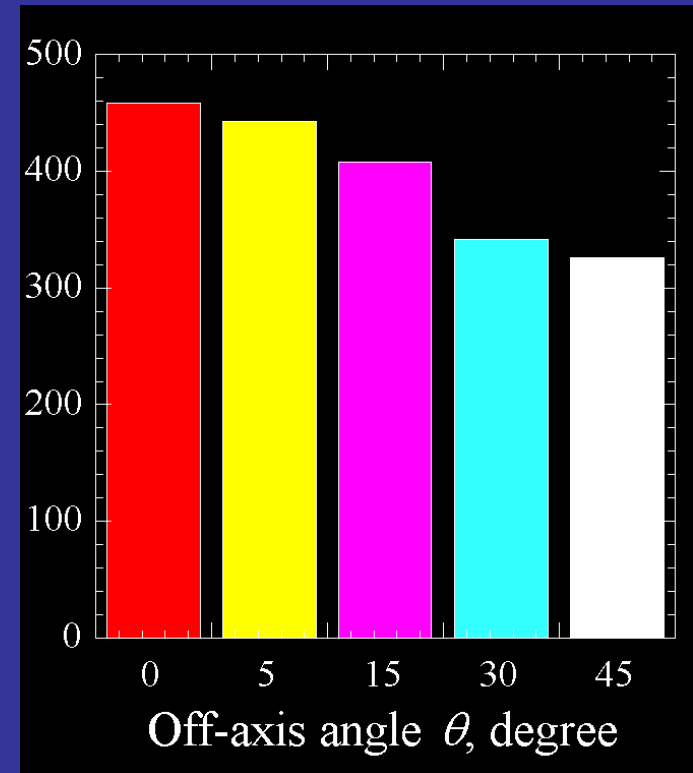


Off-Axis Stress-Strain Curves

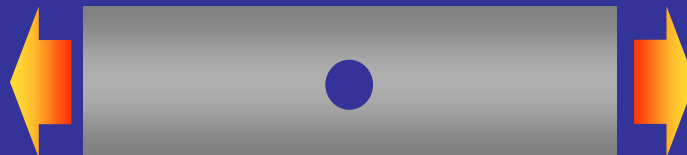
Notched behavior



Notched strength

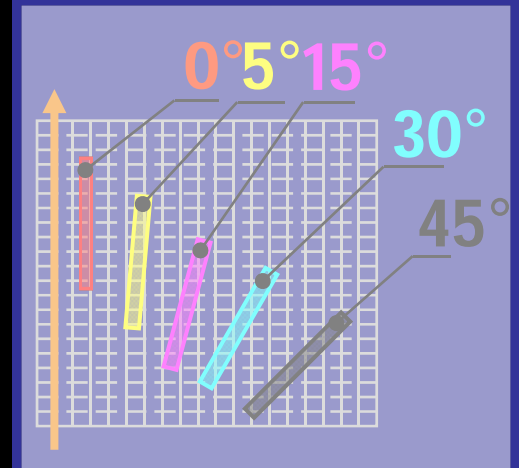
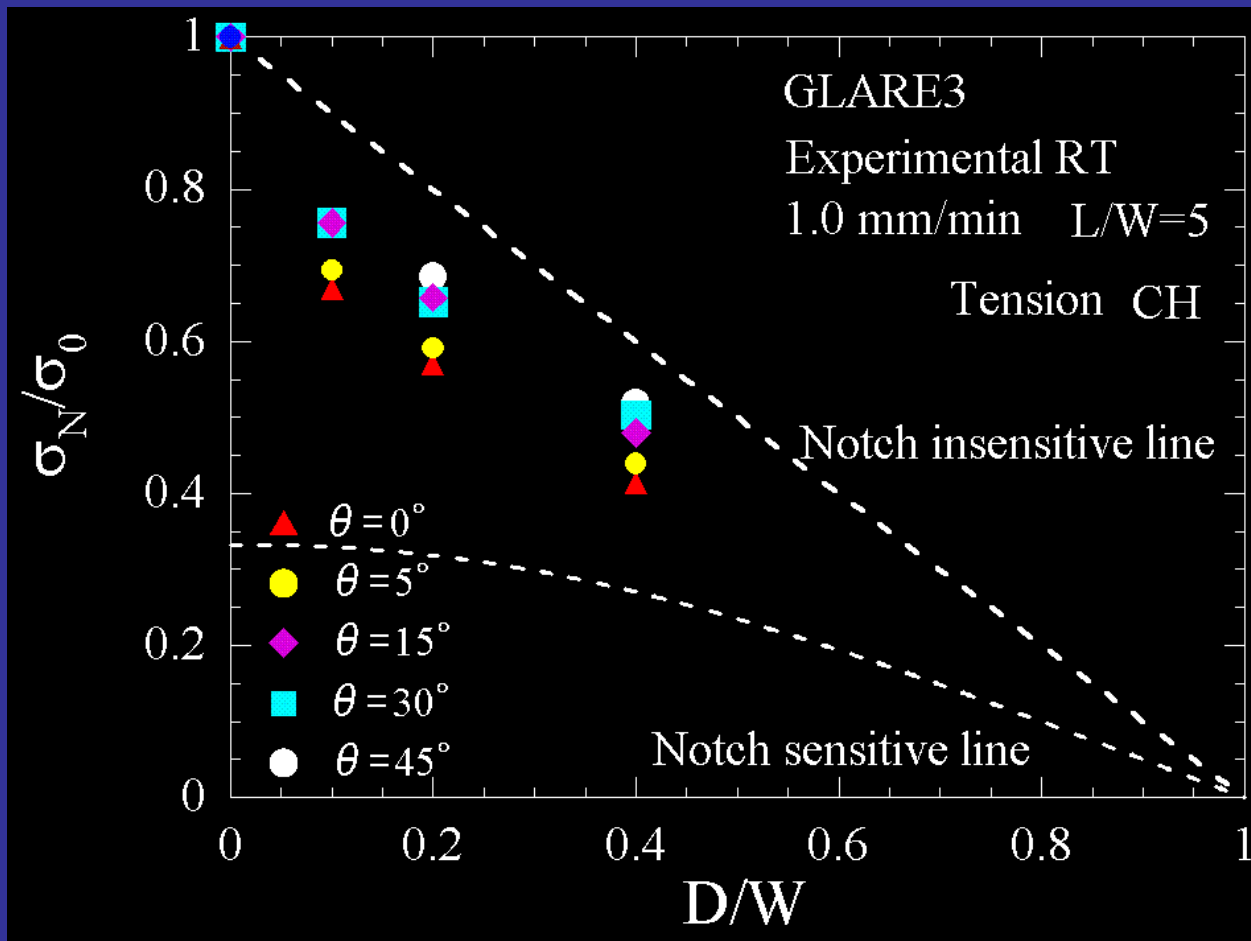
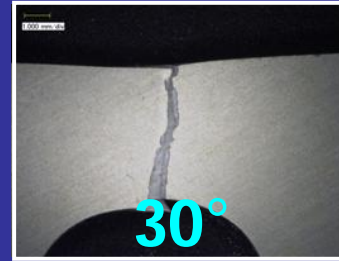
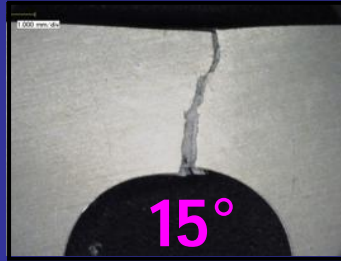


$D = 2.0$



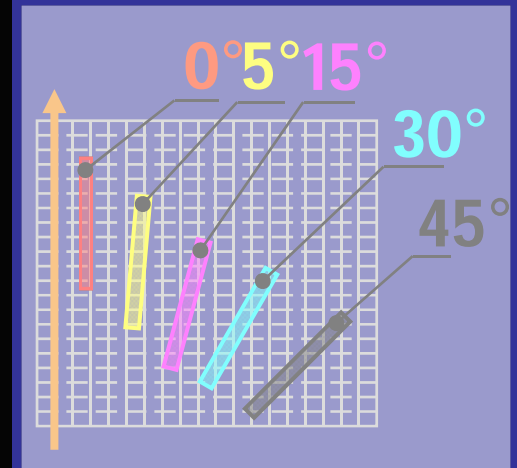
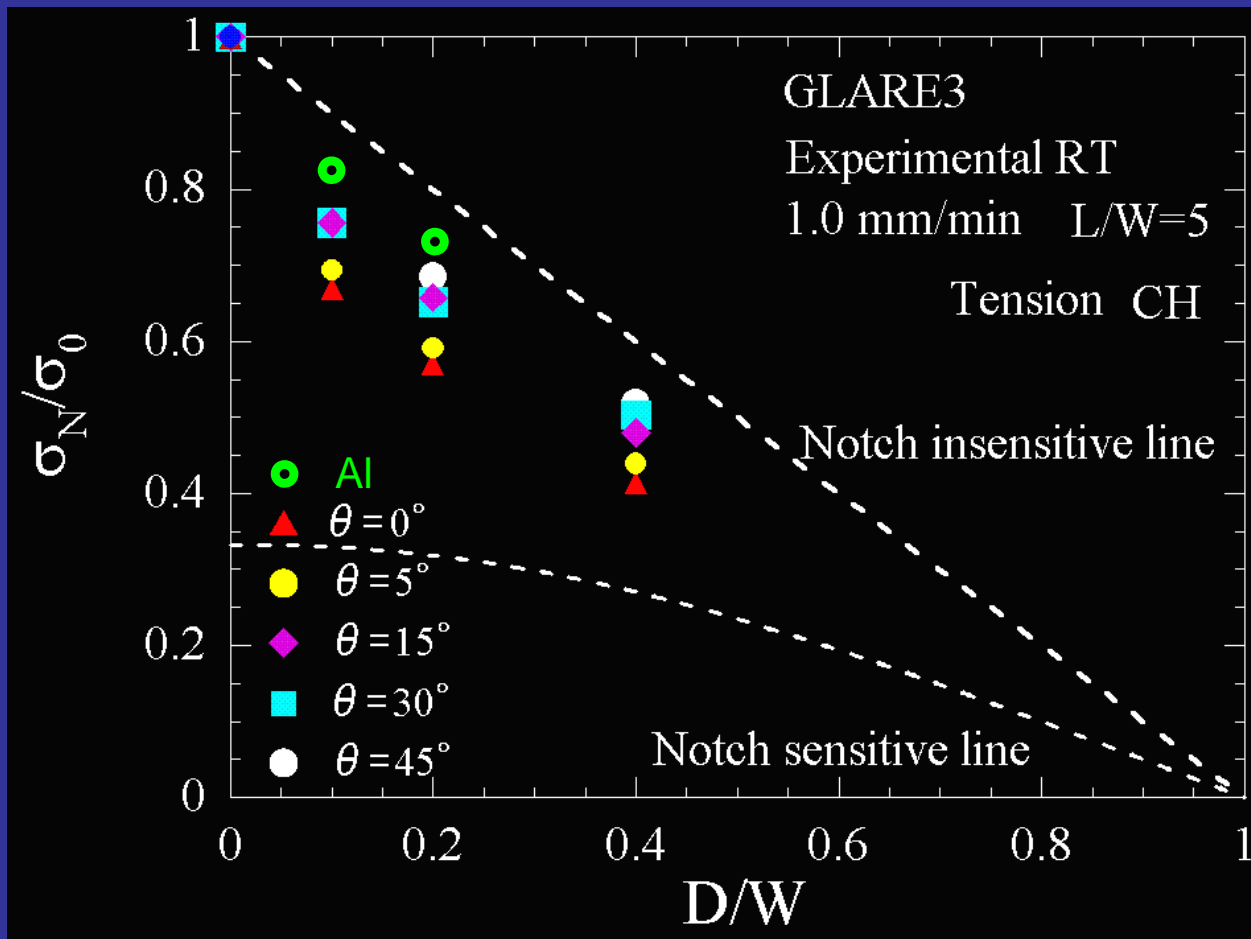
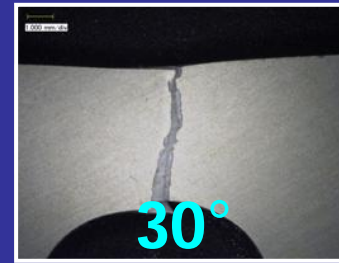
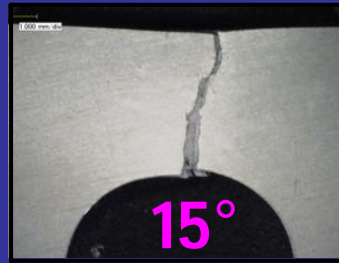


Off-Axis Notch Sensitivity Curve





Off-Axis Notch Sensitivity Curve





Application of existing fracture criteria

► **Semi-Empirical Criteria:**

Point-stress (PS) criterion

Average stress (AS) criterion

Modified PS criterion

► **Fracture Mechanics Criteria:**

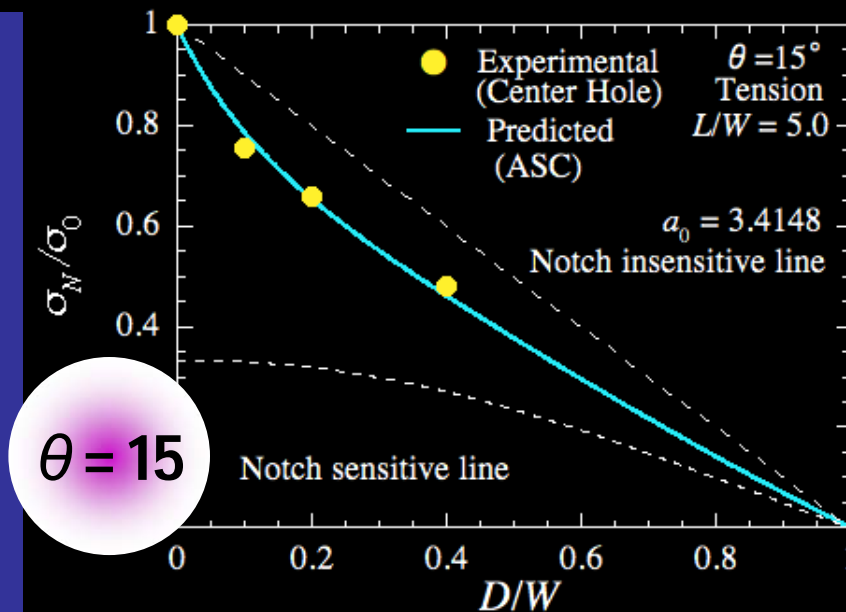
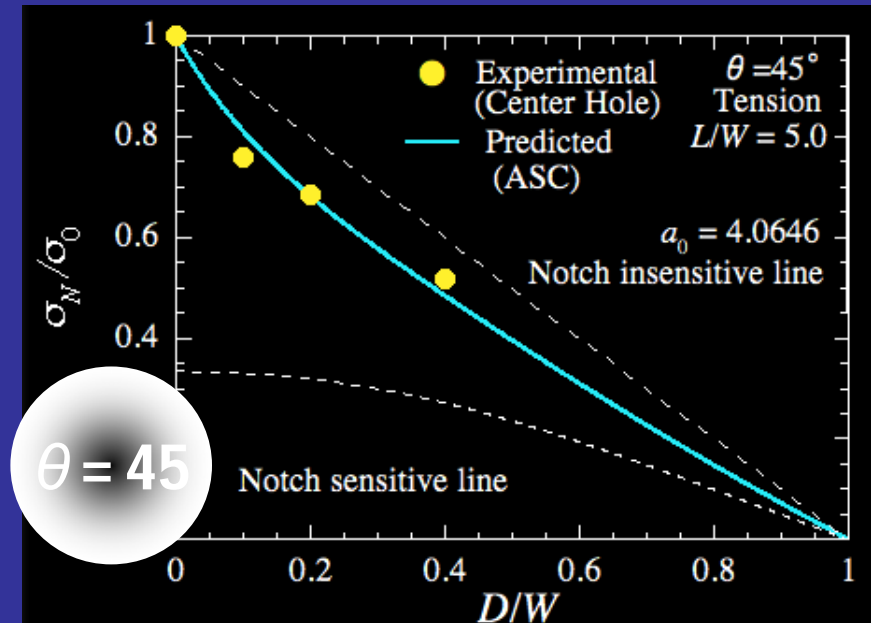
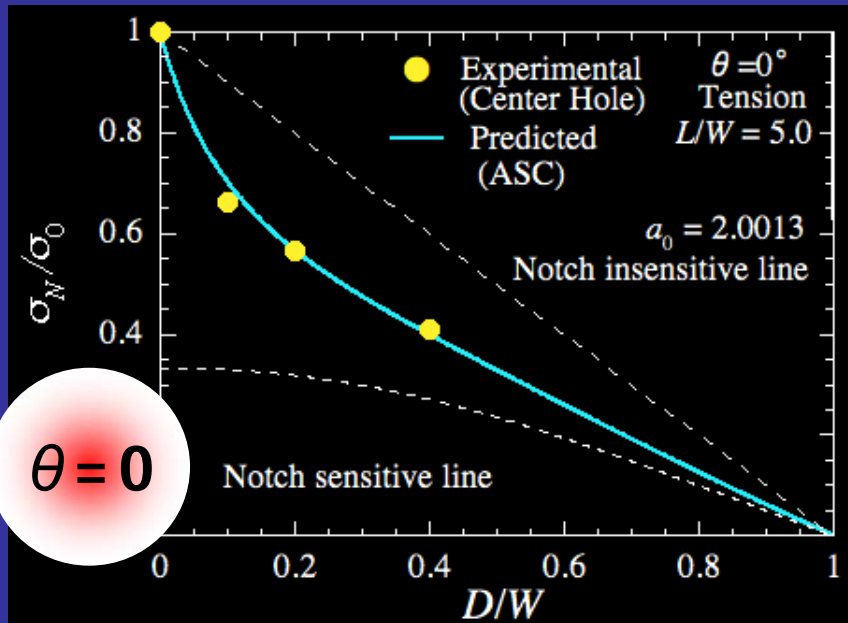
Inherent flaw criterion

Cohesive zone criterion

R-curve based criterion

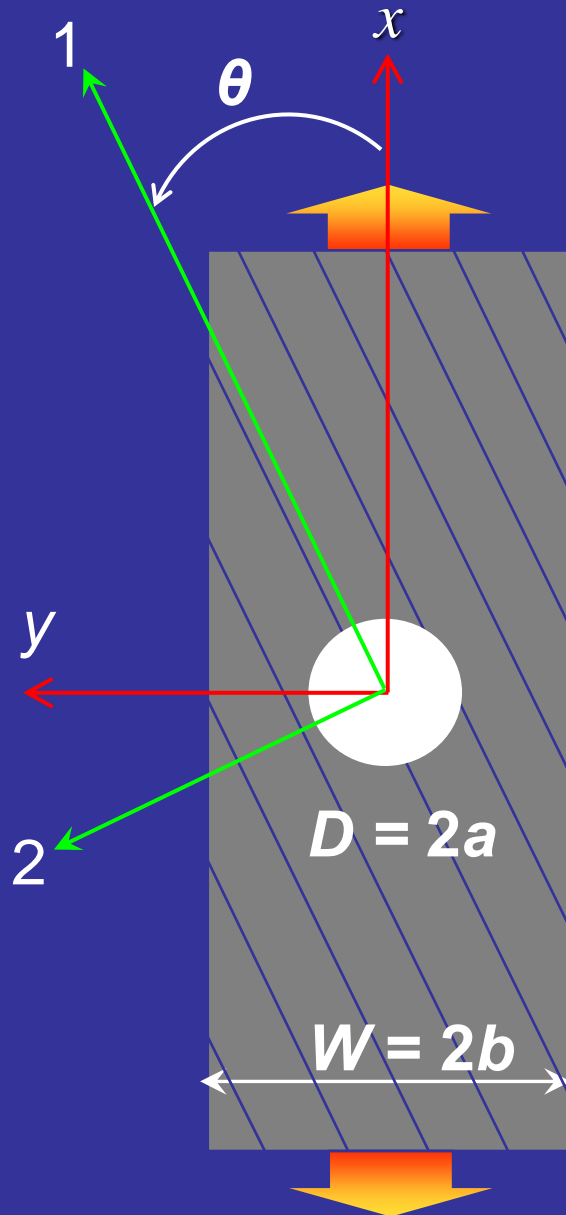


Average Stress (AS) Criterion





Off-Axis Notch Size Effect



Nominal Fracture Stress σ_N , MPa

Practical approach?

D/W

Off-axis angle θ



Multiaxial Notch Sensitivity Modeling

- ▶ A ductile failure criterion
- ▶ A brittle failure criterion
- ▶ A brittle-ductile failure criterion
- ▶ A formula for off-axis notched strength



A Ductile Failure Criterion

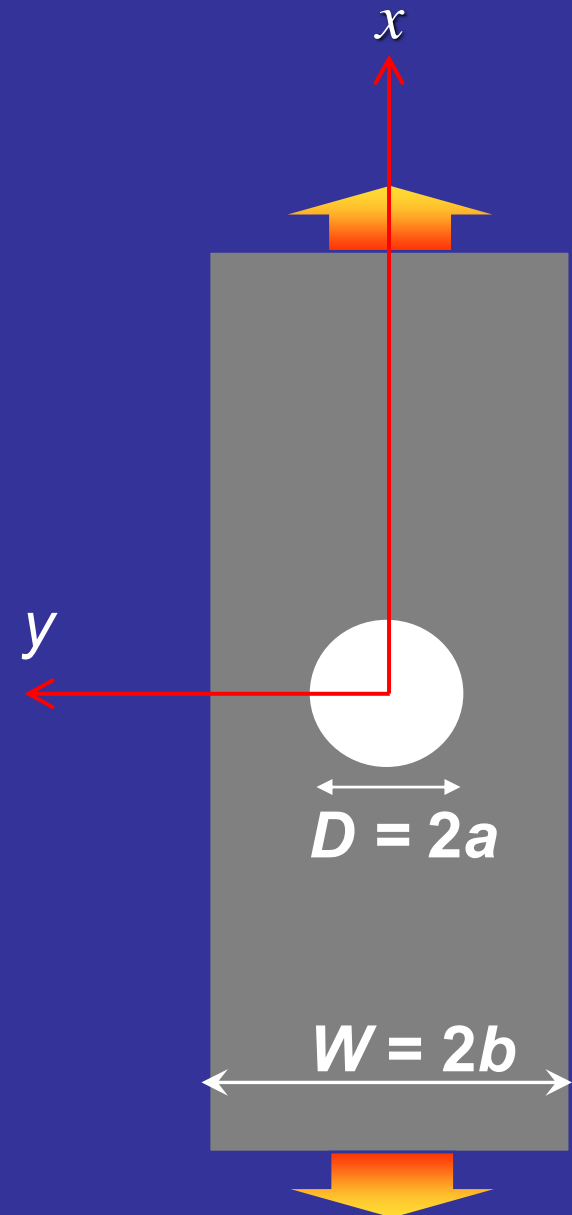
The Net Section Stress Criterion:

$$\frac{\sigma_N}{\sigma_0} = 1 - \frac{a}{b} = 1 - \alpha$$

σ_N : Remote stress at failure
(notched strength)

σ_0 : Material strength
(unnotched strength)

α : Diameter to width ratio
(= $D/W = a/b$)





A Multiaxial Ductile Failure Criterion

$$\frac{\sigma_N}{\sigma_0} = 1 - \frac{a}{b} = 1 - \alpha$$

The Net Section Stress Criterion:

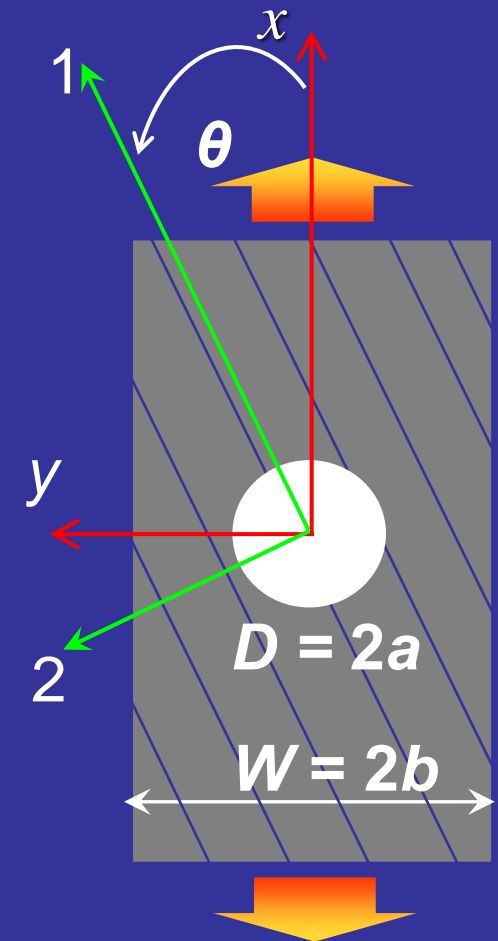
$$\frac{\sigma_{net}}{\sigma_0} = \frac{1}{1 - \alpha} \frac{\sigma_x}{\sigma_0} = 1$$



$$f_{NI} = \left(\frac{1}{1 - \alpha} \sigma^* \right)^2 = 1$$

where

$$\sigma^* = \sqrt{\left(\frac{\sigma_{11}}{X} \right)^2 - \frac{\sigma_{11}\sigma_{22}}{X^2} + \left(\frac{\sigma_{22}}{Y} \right)^2 + \left(\frac{\tau_{12}}{S} \right)^2}$$





Non-Dimensional Effective Stress

Tsai-Hill Static Failure Criterion:

$$\left(\frac{\sigma_{11}}{X}\right)^2 - \frac{\sigma_{11}\sigma_{22}}{X^2} + \left(\frac{\sigma_{22}}{Y}\right)^2 + \left(\frac{\tau_{12}}{S}\right)^2 = 1$$



X : Longitudinal strength

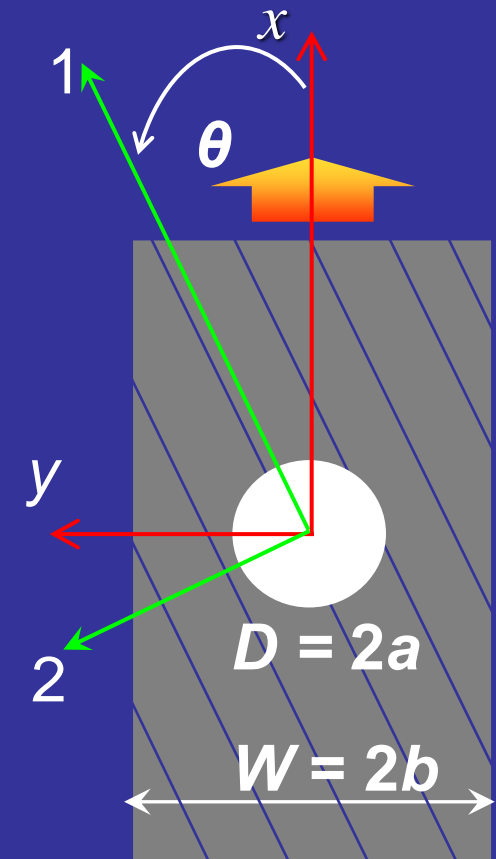
Y : Transverse strength

S : Shear strength

Non-Dimensional Effective

Stress:

$$\sigma^* \equiv \sqrt{\left(\frac{\sigma_{11}}{X}\right)^2 - \frac{\sigma_{11}\sigma_{22}}{X^2} + \left(\frac{\sigma_{22}}{Y}\right)^2 + \left(\frac{\tau_{12}}{S}\right)^2}$$





Theoretical Stress Ratio

Off-Axis Loading on UD Composites

Non-Dimensional Effective Stress

$$\sigma^* = \Omega(\theta)\sigma_x \longrightarrow \text{Static Failure Condition: } \sigma^* = \Omega(\theta)\sigma_x = 1$$

$$\therefore \sigma_0 = \frac{1}{\Omega(\theta)}$$

$$\text{where } \Omega(\theta) = \frac{1}{\sqrt{\left(\frac{\cos^2 \theta}{X}\right)^2 - \frac{\cos^2 \theta \sin^2 \theta}{X^2} + \left(\frac{\sin^2 \theta}{Y}\right)^2 + \left(\frac{-\cos \theta \sin \theta}{S}\right)^2}}$$

Theoretical stress ratio for off-axis loading:

$$\sigma^* = \Omega(\theta)\sigma_x = \frac{\sigma_x}{\frac{1}{\Omega(\theta)}} = \frac{\sigma_x}{\sigma_0}$$



A Brittle Failure Criterion

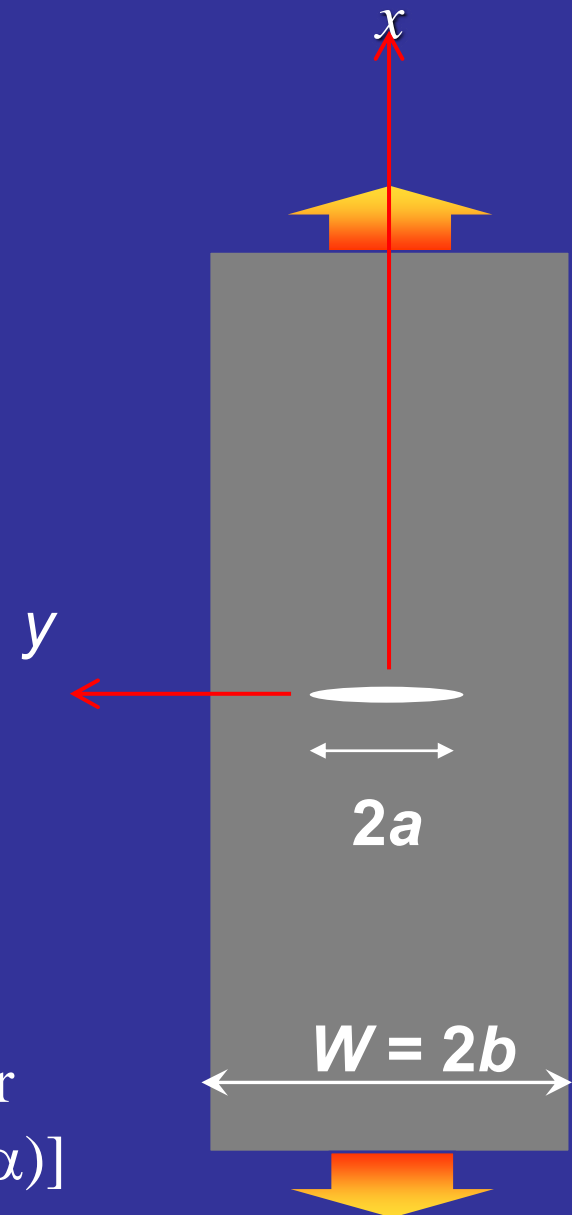
Fracture Mechanics Criterion:

$$\frac{\sigma_N}{\sigma_0} = \frac{K_{IC}}{F \sigma_0 \sqrt{\pi a}}$$

σ_N : Remote stress at failure
(notched strength)

σ_0 : Material strength
(unnotched strength)

F : Finite width correction factor
[$F = F(2a/W) = F(a/b) = F(\alpha)$]



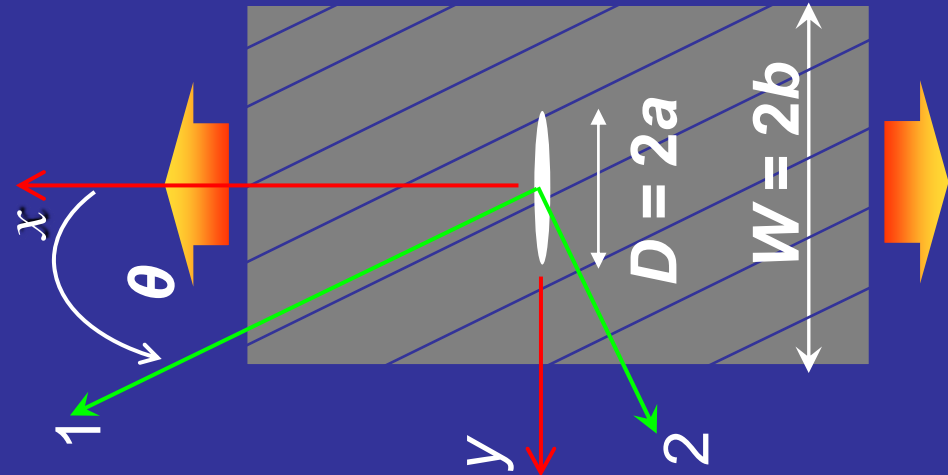


A Multiaxial Brittle Failure Criterion

$$\frac{\sigma_N}{\sigma_0} = \frac{K_{IC}}{\sigma_0 F \sqrt{\pi a}}$$

Fracture Mechanics Criterion:

$$\frac{K_I}{K_{IC}} = \frac{\sigma_x F \sqrt{\pi a}}{K_{IC}} = 1$$



$$f_{NS} = f_{NS}(k_{11}, k_{22}, k_{12}) = \left(\frac{k_{11}}{K_{IC}^{11}} \right)^2 - \frac{k_{11} k_{22}}{(K_{IC}^{11})^2} + \left(\frac{k_{22}}{K_{IC}^{22}} \right)^2 + \left(\frac{k_{12}}{K_{IC}^{12}} \right)^2 = 1$$

where

$$\begin{pmatrix} k_{11} \\ k_{22} \\ k_{12} \end{pmatrix} = F_I \sqrt{\pi a} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & \chi \end{bmatrix} \begin{pmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{12} \end{pmatrix}$$



A Multiaxial Brittle Failure Criterion

Normalized principal fracture toughness

(Principal stress brittleness numbers)

$$\frac{K_{IC}^{11}}{X\sqrt{b}} \equiv \kappa_{11}$$

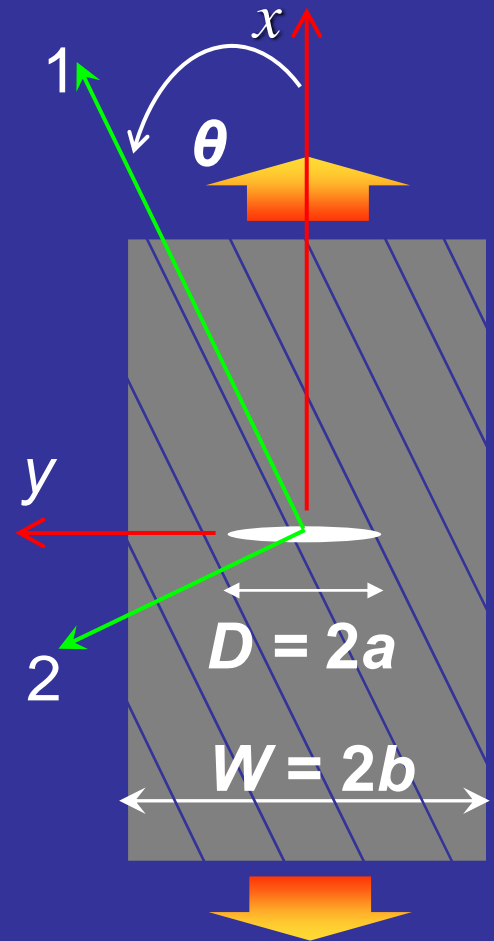
$$\frac{K_{IC}^{22}}{Y\sqrt{b}} \equiv \kappa_{22}$$

$$\frac{K_{IC}^{12}}{S\sqrt{b}} \equiv \kappa_{12}$$

extension

$$s = \frac{K_{IC}}{\sigma_u \sqrt{b}}$$

(Carpinteri)



$$f_{NS} = F_I^2 \pi \alpha \left[\left(\frac{\sigma_{11}}{X\kappa_{11}} \right)^2 - \frac{\sigma_{11}\sigma_{22}}{(X\kappa_{11})^2} + \left(\frac{\sigma_{22}}{Y\kappa_{22}} \right)^2 + \chi^2 \left(\frac{\sigma_{12}}{S\kappa_{12}} \right)^2 \right]$$



A Multiaxial Brittle-Ductile Failure Criterion

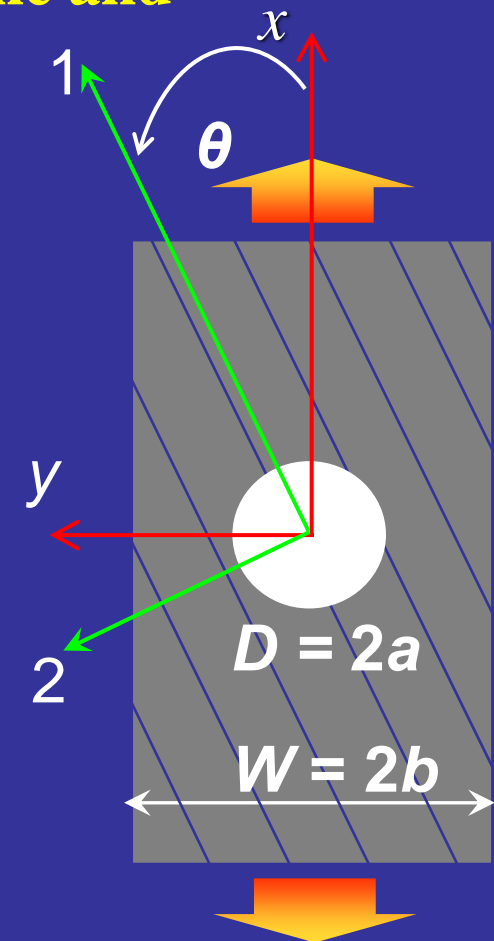
A multiaxial failure criterion for orthotropic solids with any notch sensitivity bounded by the ductile and brittle limits:

$$f = f_{NI} + f_{NS} = 1$$

where

$$f_{NI} = \left(\frac{\sigma^*}{1 - \alpha} \right)^2$$

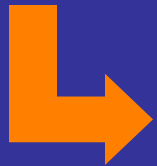
$$f_{NS} = F_I^2 \pi \alpha \left[\left(\frac{\sigma_{11}}{X\kappa_{11}} \right)^2 - \frac{\sigma_{11}\sigma_{22}}{(X\kappa_{11})^2} + \left(\frac{\sigma_{22}}{Y\kappa_{22}} \right)^2 + \chi^2 \left(\frac{\sigma_{12}}{S\kappa_{12}} \right)^2 \right]$$





A Formula for Off-Axis Notched Strength

$$f = f_{NI} + f_{NS} = 1$$

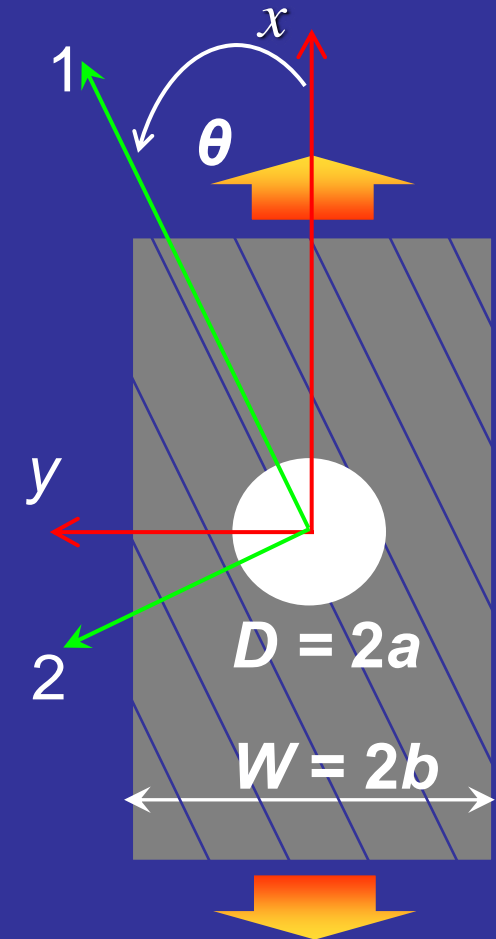


$$\frac{\sigma_N}{\sigma_0} = \frac{1}{\sqrt{\left(\frac{1}{1-\alpha}\right)^2 + \left(\frac{\Omega_{NS}}{\Omega_0}\right)^2}}$$

where

$$\Omega_0 = \sqrt{\frac{m^4}{X^2} - \frac{m^2 n^2}{X^2} + \frac{n^4}{Y^2} + \frac{m^2 n^2}{S^2}} = \frac{1}{\sigma_0}$$

$$\Omega_{NS} = \sqrt{F_I^2 \pi \alpha \left\{ \frac{m^4 - m^2 n^2}{(X\kappa_{11})^2} + \frac{n^4}{(Y\kappa_{22})^2} + \chi^2 \frac{m^2 n^2}{(S\kappa_{12})^2} \right\}}$$



$$\begin{cases} m = \cos \theta \\ n = \sin \theta \end{cases}$$

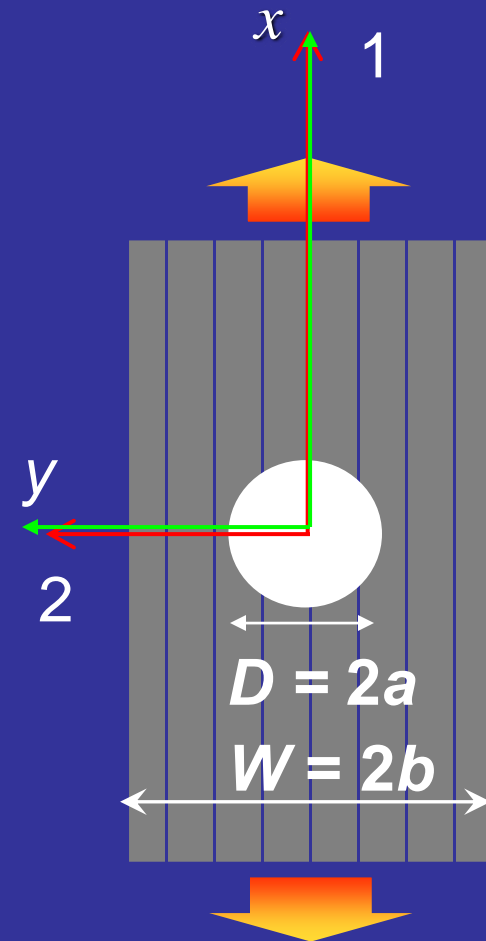


Principal Notched Strengths

$$\frac{X_N}{X} = \frac{1}{\sqrt{\frac{1}{(1-\alpha)^2} + \frac{F_I^2 \pi \alpha}{\kappa_{11}^2}}}$$

$$\frac{Y_N}{Y} = \frac{1}{\sqrt{\frac{1}{(1-\alpha)^2} + \frac{F_I^2 \pi \alpha}{\kappa_{22}^2}}}$$

$$\frac{S_N}{S} = \frac{1}{\sqrt{\frac{1}{(1-\alpha)^2} + \chi^2 \frac{F_I^2 \pi \alpha}{\kappa_{12}^2}}}$$



(a generalization of the Suo-Ho-Gong model, 1993)



Verification



Application to GLARE-3

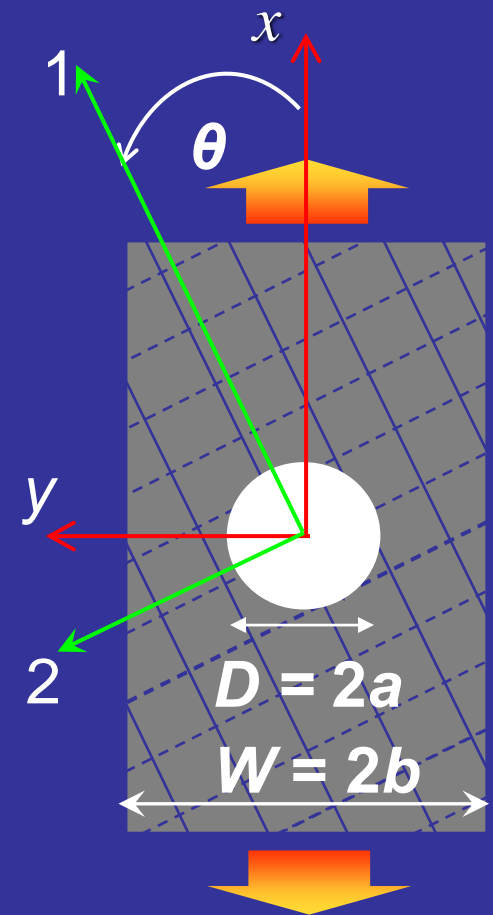
$$\frac{X_N}{X} = \frac{1}{\sqrt{\frac{1}{(1-\alpha)^2} + \frac{F_I^2 \pi \alpha}{\kappa_I^2}}} = \frac{Y_N}{Y}$$

$$\frac{S_N}{S} = \frac{1}{\sqrt{\frac{1}{(1-\alpha)^2} + \chi^2 \frac{F_I^2 \pi \alpha}{\kappa_{II}^2}}}$$

where

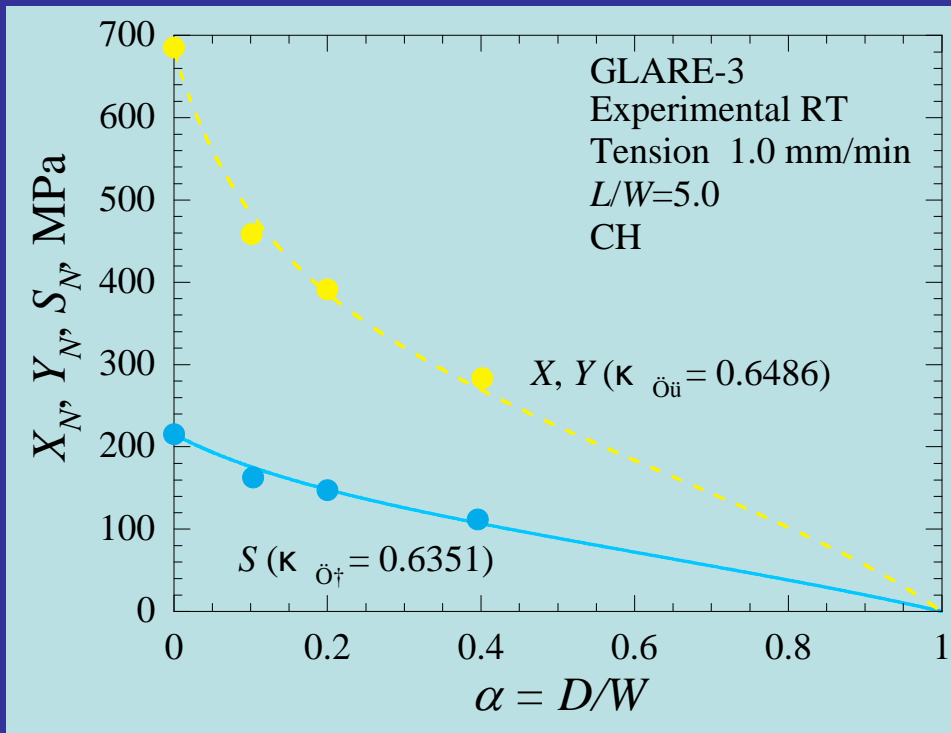
$$F_I = \left(1 - 0.025\alpha^2 + 0.06\alpha^4\right) \sqrt{\sec \frac{\pi\alpha}{2}}$$

$$\chi \approx 0.6 \quad (\text{Sih and Chen, 1973})$$





Principal Notch Sensitivity in GLARE-3



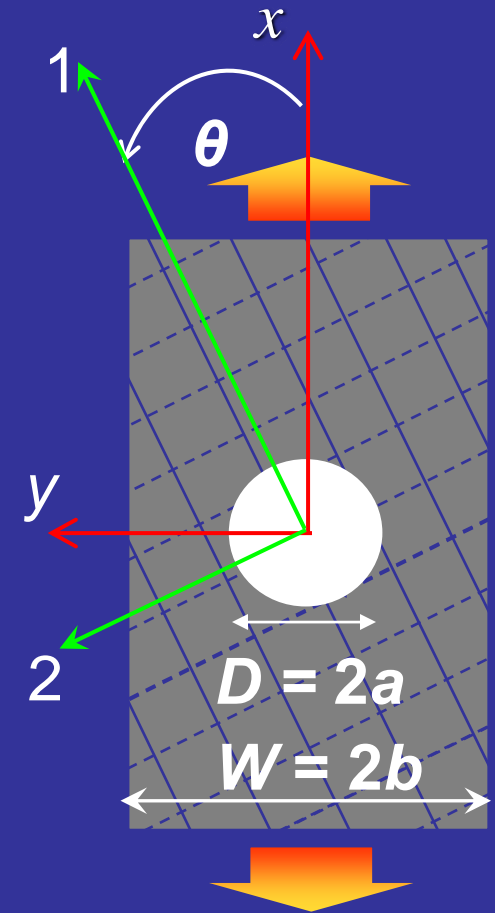
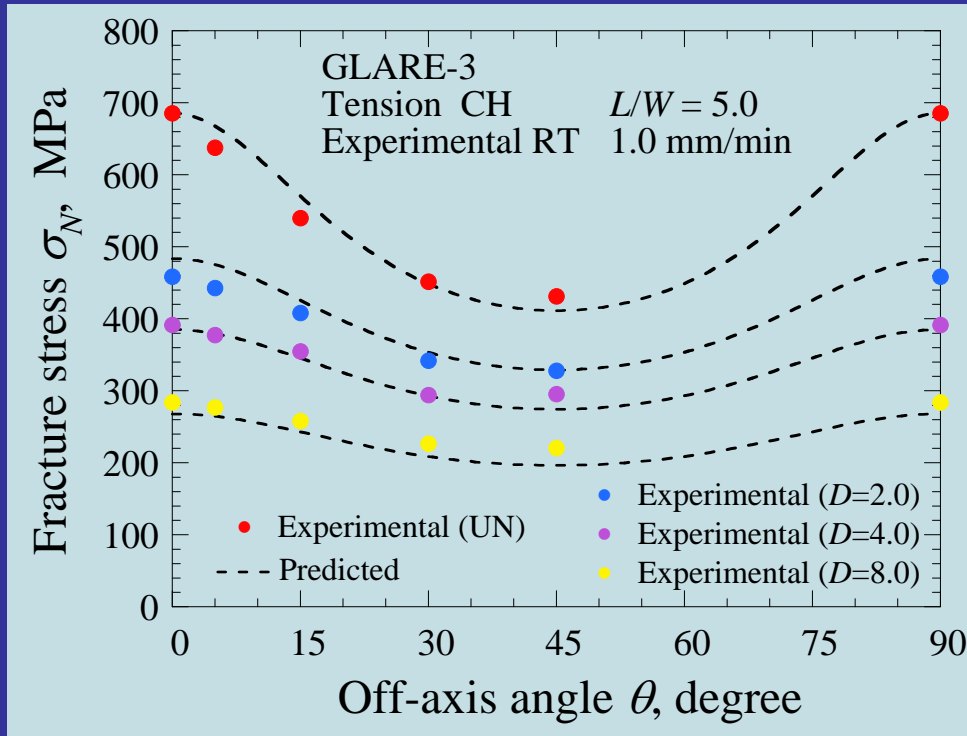
$$\frac{X_N}{X} = \frac{1}{\sqrt{\frac{1}{(1-\alpha)^2} + \frac{F_I^2 \pi \alpha}{\kappa_I^2}}} = \frac{Y_N}{Y}$$

$$\frac{S_N}{S} = \frac{1}{\sqrt{\frac{1}{(1-\alpha)^2} + \chi^2 \frac{F_I^2 \pi \alpha}{\kappa_{II}^2}}}$$



Off-Axis Notched Strength of GLARE-3

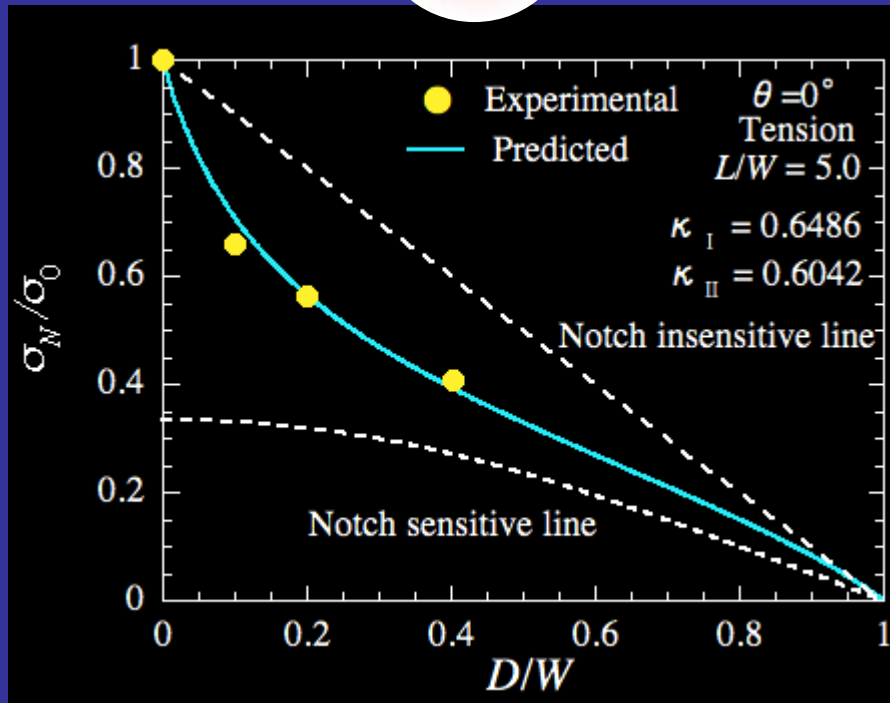
Predicted



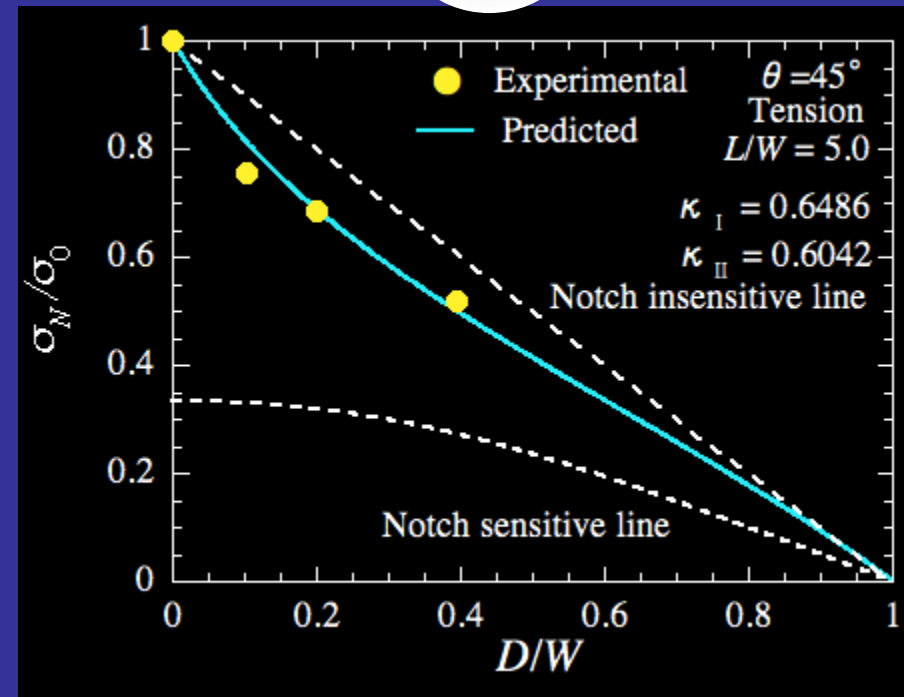


Predicted Off-Axis Notch Sensitivity in GLARE-3

$\theta = 0$



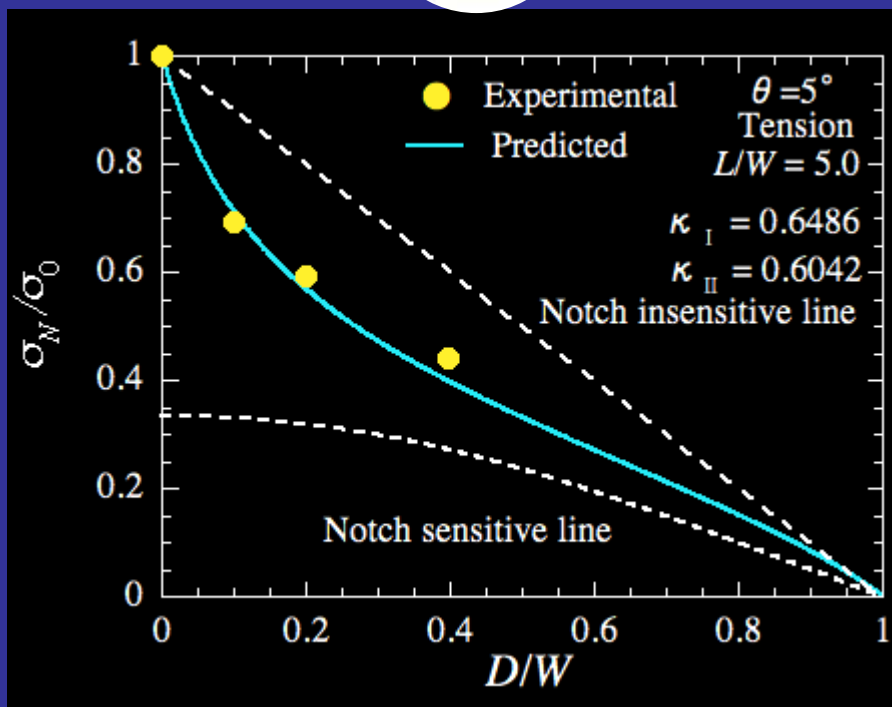
$\theta = 45$



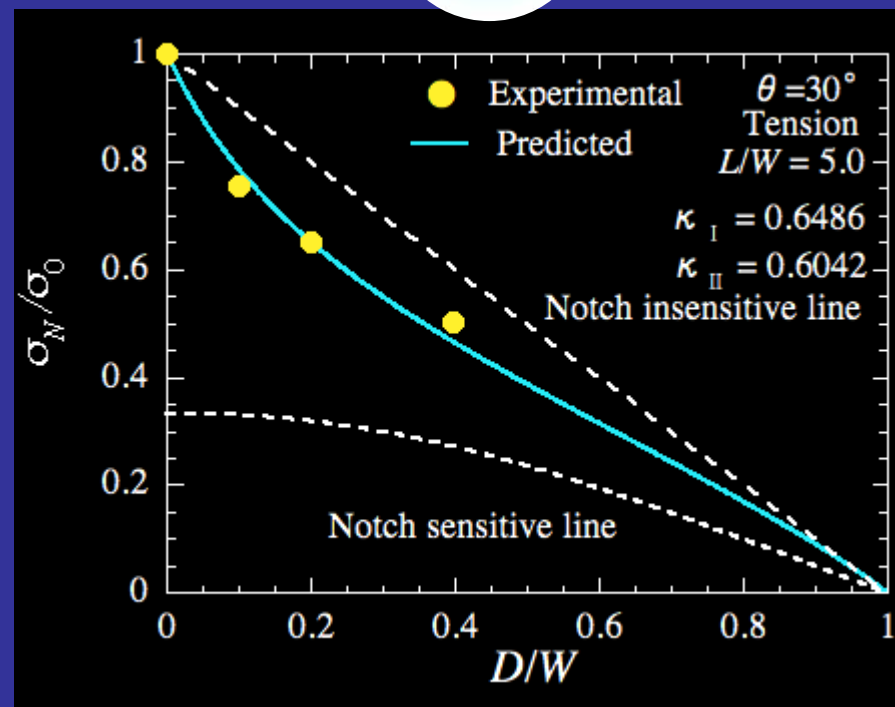


Predicted Off-Axis Notch Sensitivity in GLARE-3

$\theta = 5^\circ$



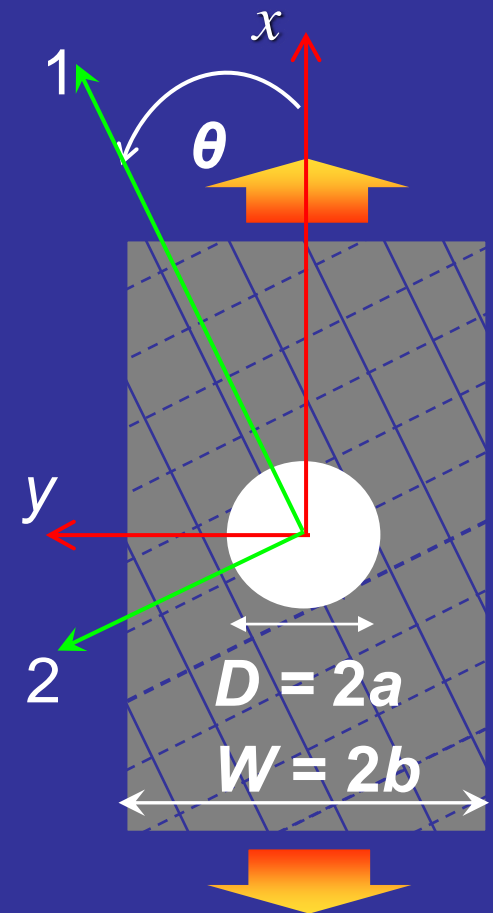
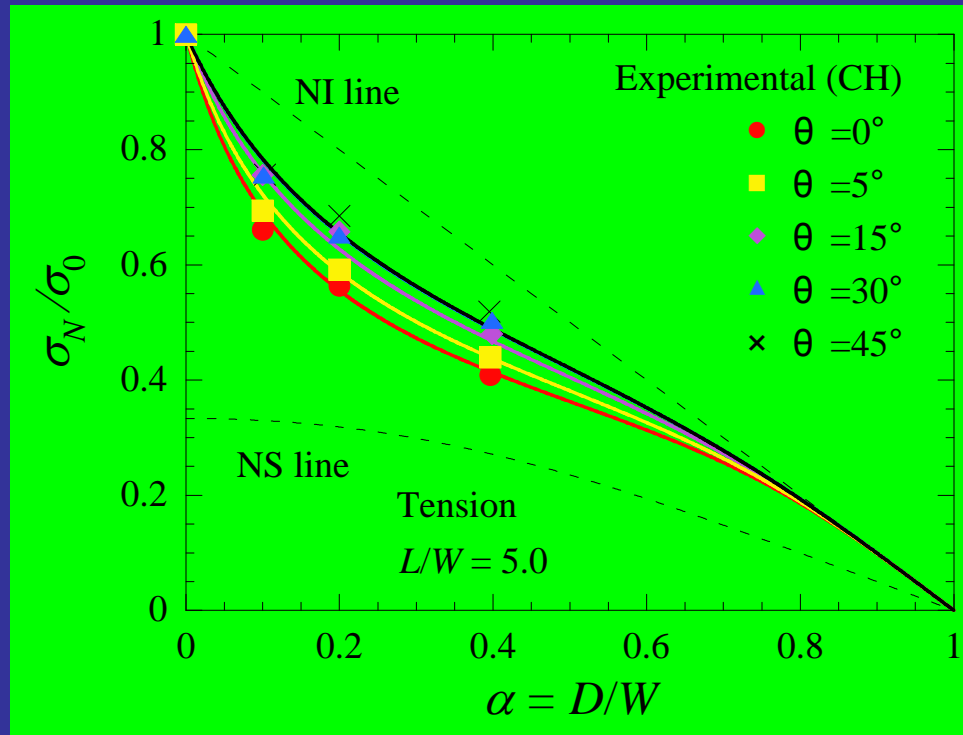
$\theta = 30^\circ$





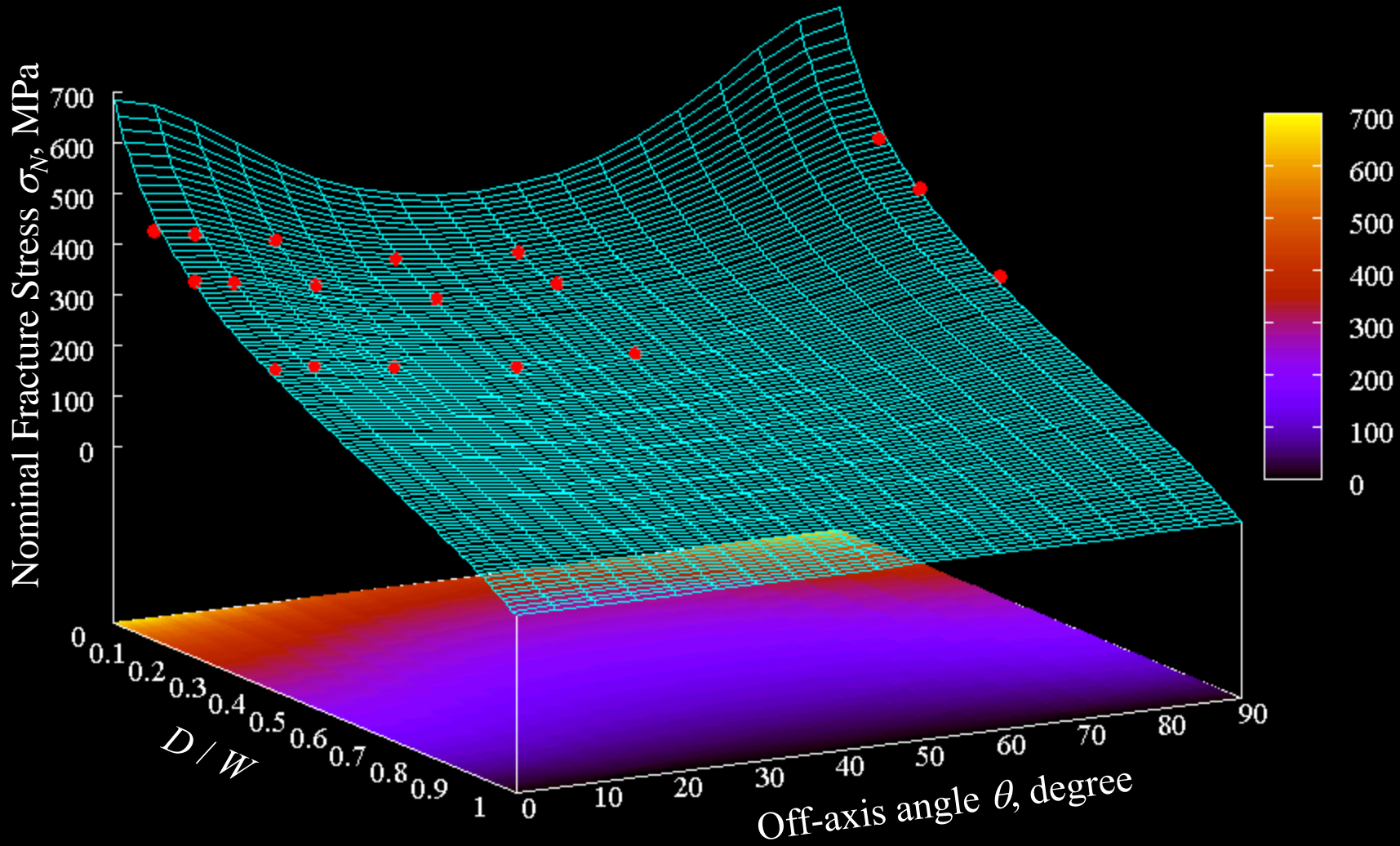
Fiber Orientation Dependence of Off-Axis Notch Sensitivity in GLARE-3

Predicted



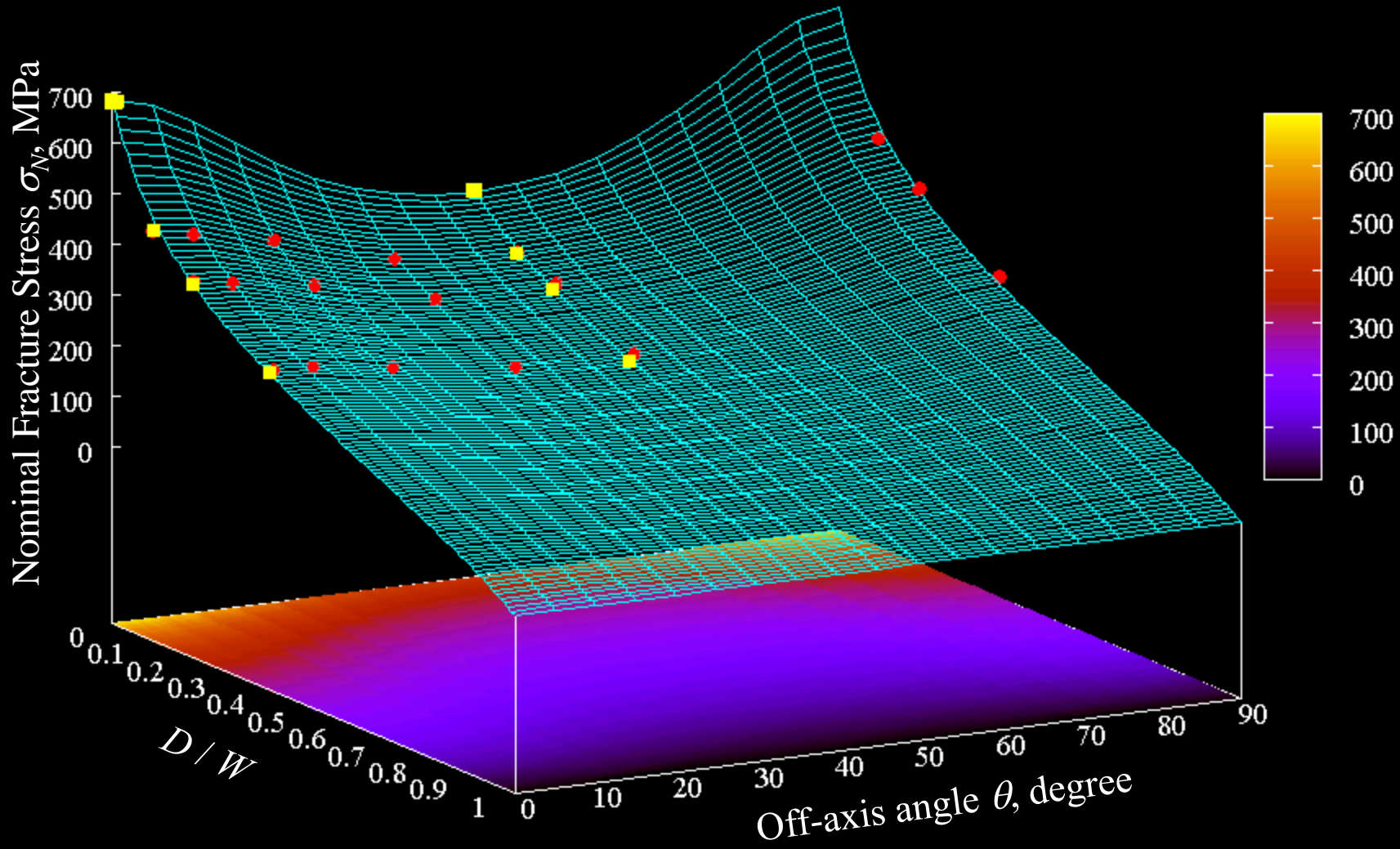


Usefulness of the Formula





Usefulness of the Formula





Conclusions (1/2)

Experimental study was conducted on the off-axis specimens with central open circular holes for different notch sizes and fiber orientations in order to identify the notch sensitivity in GLARE-3. **Theoretical** attempt at general formulation of a failure criterion applicable to notched orthotropic composites was also made.

EXPERIMENTAL

- The notched strength of Glare-3 decreases with increasing notch dimension, regardless of the fiber orientation.
- The off-axis notched strength of Glare-3 is bounded by the notch insensitive and sensitive limits, regardless of the fiber orientation, indicating moderate sensitivity to a notch.
- The notch sensitivity of Glare-3 depends on the fiber orientation of the GFRP layers, and the notch sensitivity of Glare-3 is highest in the fiber direction and lowest in the 45° direction.



Conclusions (2/2)

THEORETICAL

- A new multiaxial failure criterion for notched orthotropic materials with any notch sensitivity was developed.
- It was formulated by combining a notch insensitive (ductile) and sensitive (brittle) failure criteria based on the net section stress criterion and the fracture mechanics criterion, respectively.
- A formula was derived to efficiently predict the off-axis notched strength of orthotropic fiber reinforced composites for any length of a notch as well as for any fiber orientation.
- The proposed failure criterion succeeded in accurately and efficiently describing the notch size effect and the fiber orientation dependence of the off-axis notched strength of the fiber metal laminate GLARE-3.

**Thank you for your kind
attention !**